

# **SPICAM on Mars Express: an overview of near IR spectrometer of SPICAM and its results.**

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# SPICAM Light / MEX



IR: 1-1.7  $\mu\text{m}$   
AOTF  
spectrometer  
Two pixels



+Z (nadir)

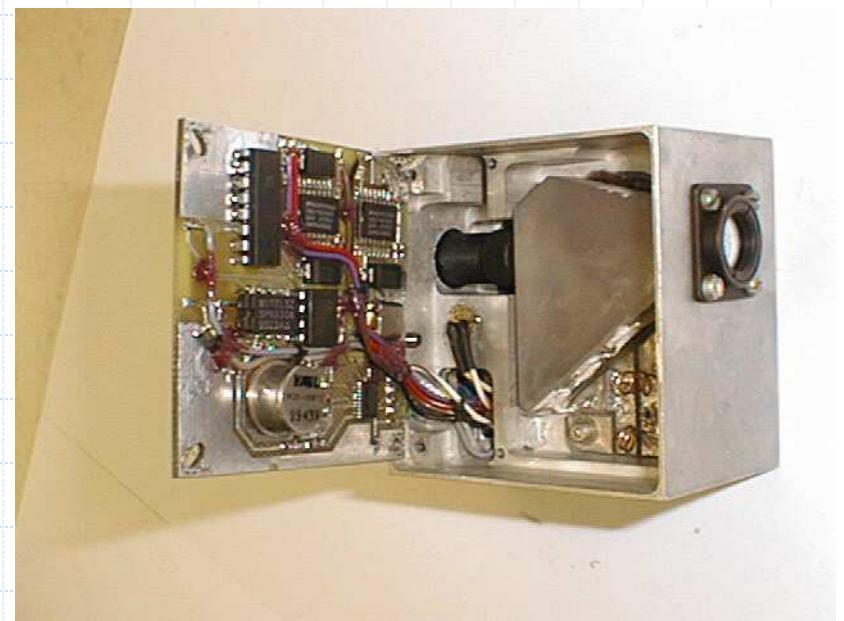
UV:  
110-310 nm  
spherical  
grating  
imaging  
spectrometer  
with slit  
mechanism

# History of the AOTF on MEX

- ◆ SPICAM/Mars 96 originally consisted of stellar par (15 kg at the platform) and solar part (19 kg). It was not in the model payload of Mars Express
- ◆ 1997 AO: proposed UV spectrometer and IR solar occultation/nadir channel (~10 kg total); after peer review only UV was chosen (5 kg)
- ◆ Descoped IR reproposed for solar occultation only (~2.5 kg) without success.
- ◆ A concept of AOTF Nadir channel (1 kg) proposed
- ◆ 1999: prototype Nadir channel demonstrated and accepted by ESA an add-on to the UV optical block with **no additional mass allowed**

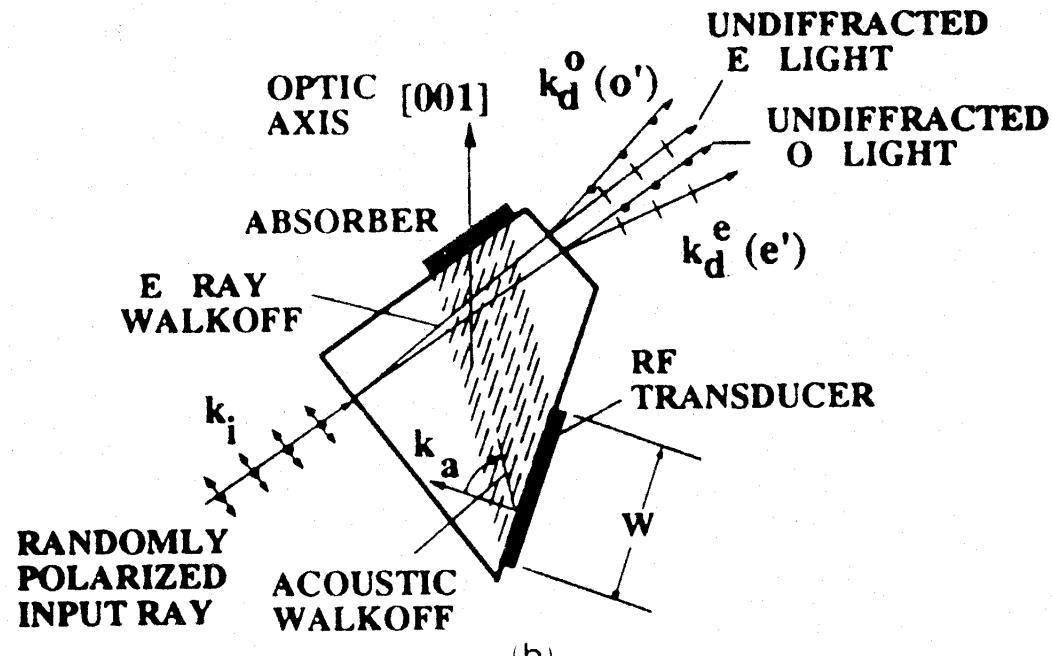
# First Acousto Optic Optic Tuneable Filter (AOTF) ever flown in civil space

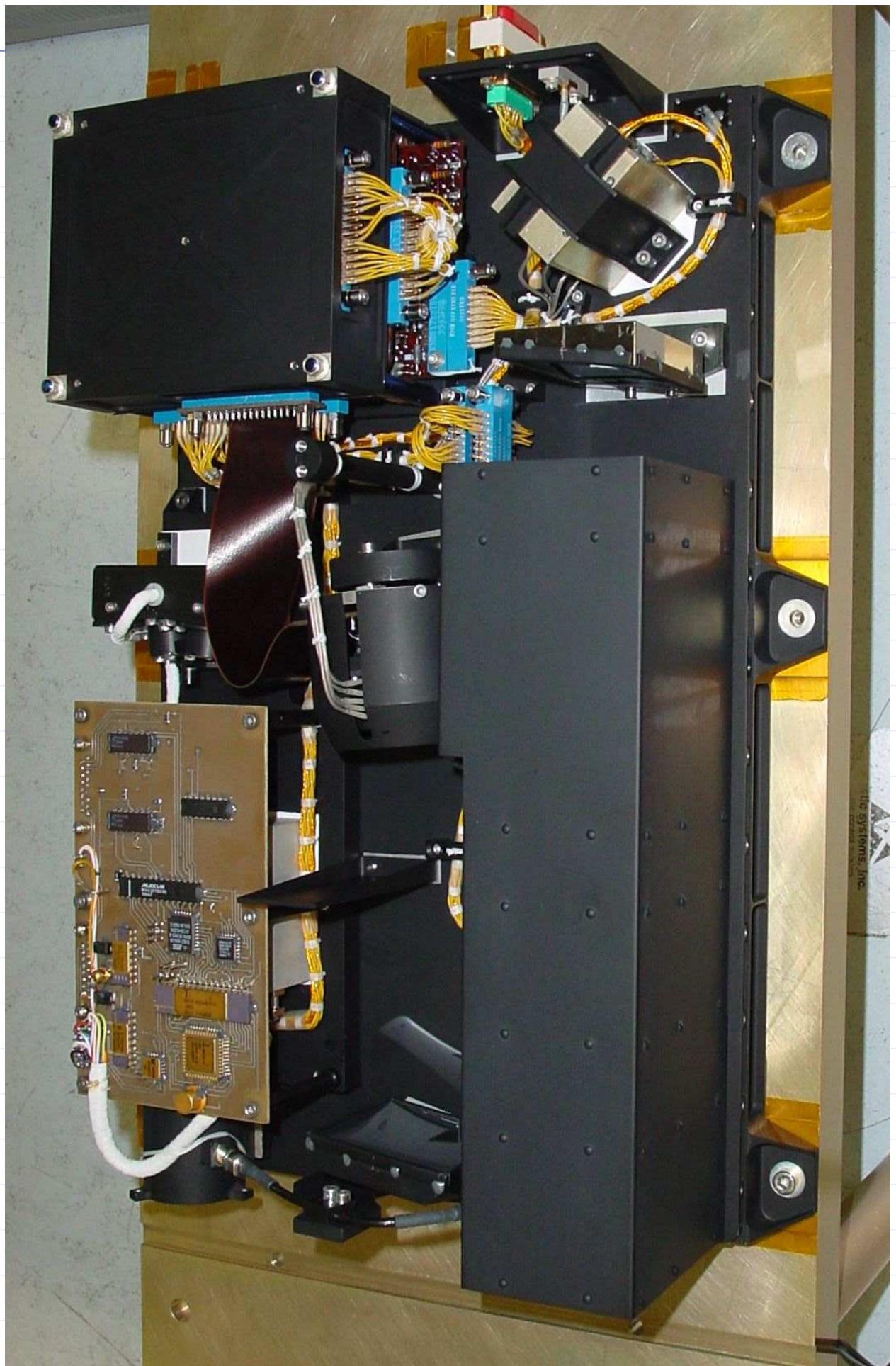
- ◆ Mass inferior to 1 kg (700 g, DC/DC and Solar entry not included)
- ◆ Spectral resolution specified as  $3.5 \text{ cm}^{-1}$  ( $\lambda/\Delta\lambda > 1300$ )
- ◆ Capable of measuring  $\text{H}_2\text{O}$  in Mars atmosphere similar to MAWD



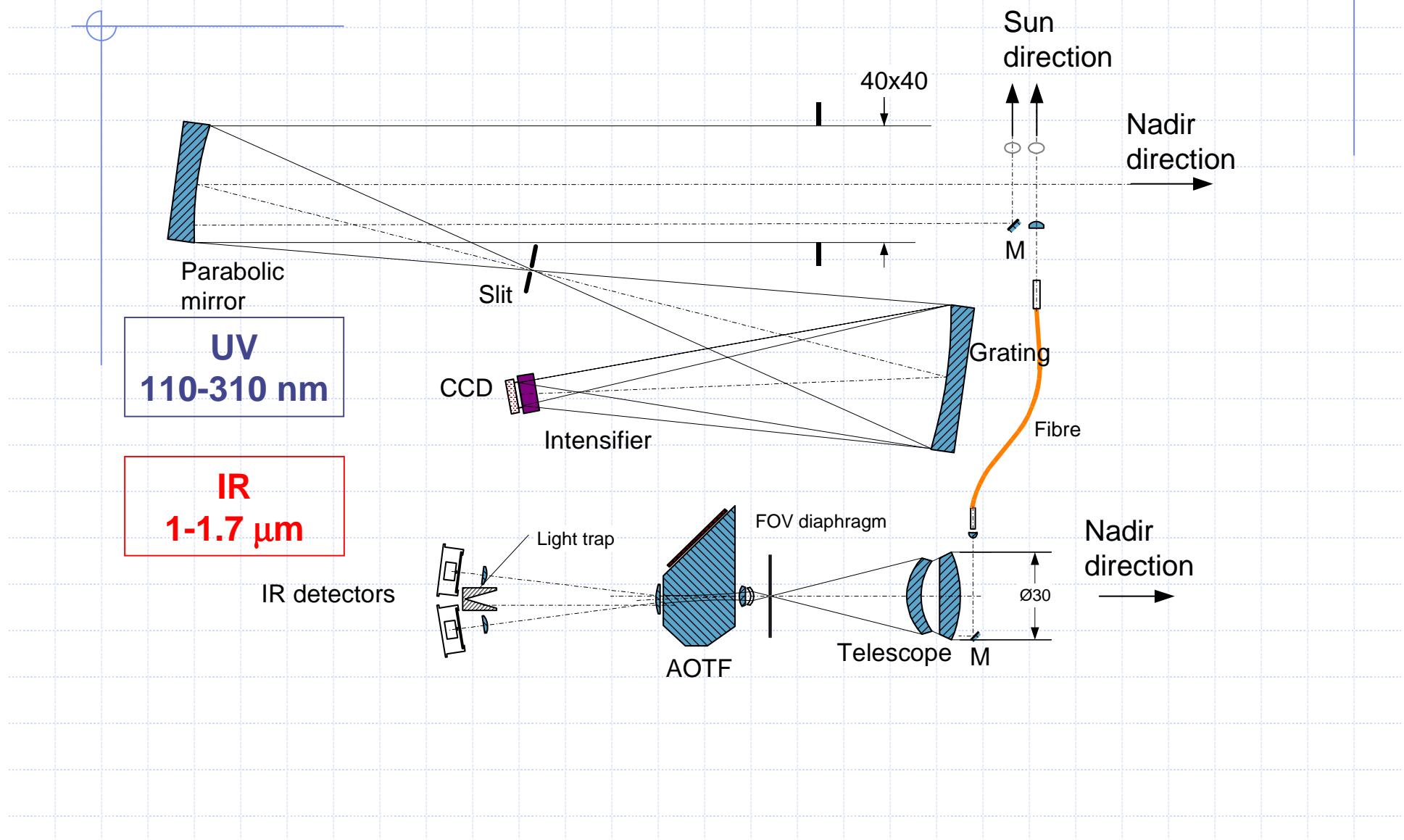
# Acousto Optic Filtration

- ◆ TeO<sub>2</sub> birefringent crystal (0.4-5.2 μm)      1-1.7 μm
- ◆ Efficiency < 60-70%      ~25%
- ◆ Max resolving power ~1500      >1700
- ◆ Aperture < 1 cm<sup>2</sup>, < 5-6°      1:1.5
- ◆ Imaging      not used
- ◆ Spectral range x2 and more\*
- ◆ RF ~ 10-200 MHz
- ◆ Power consumption
- ◆ Time to change λ





# SPICAM optical scheme



# Modes of operation

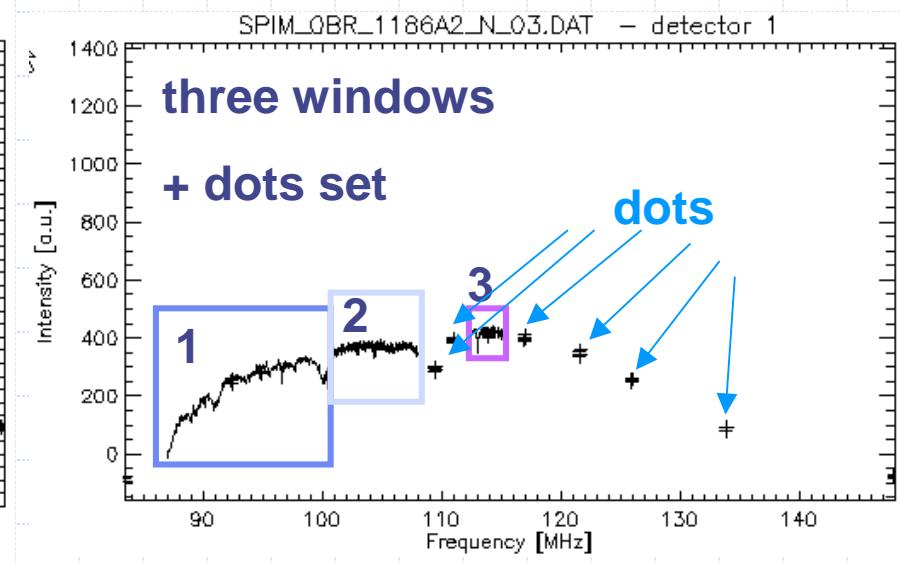
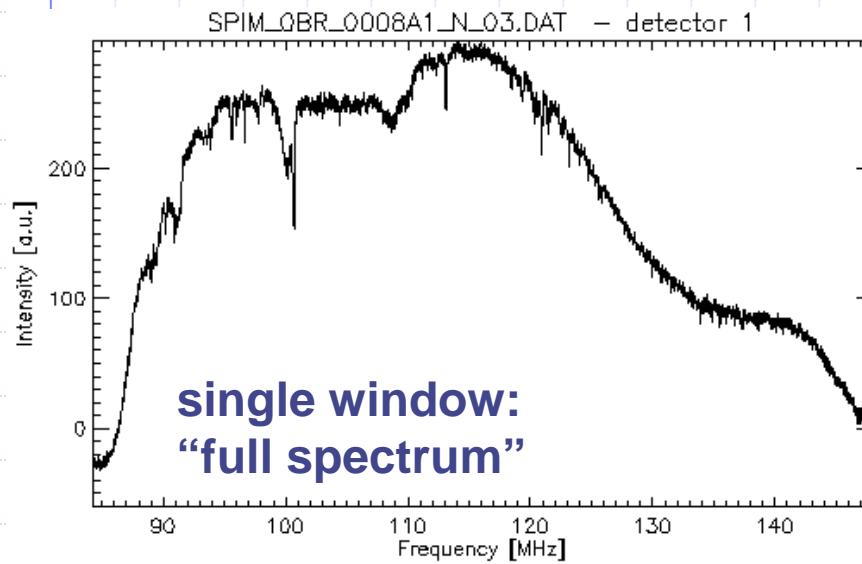
**IR channel** = 2 detectors ( $\neq$  polarization)

Spectra acquisition in 1, 2 or 3 "windows"

+ dots set (starting frequency, points, step)

max points = 3984, max acquisition time = 24s)

normally points = 664-1328, acquisition time = 2-4s



# Calibrations

## ◆ Radiometric calibration

- Done by comparison with OMEGA data and using high-resolution solar spectrum

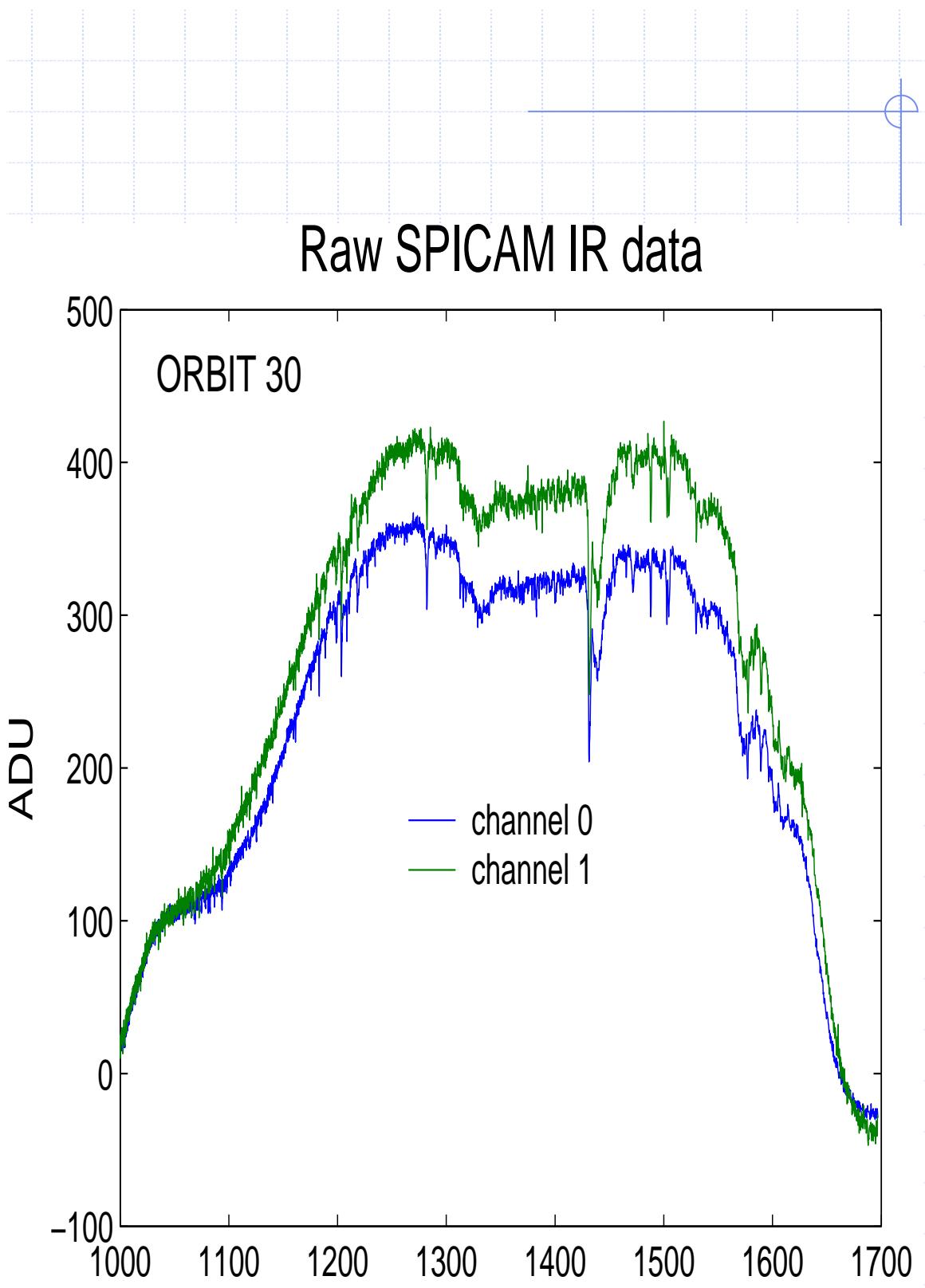
## ◆ Wavelength assignment

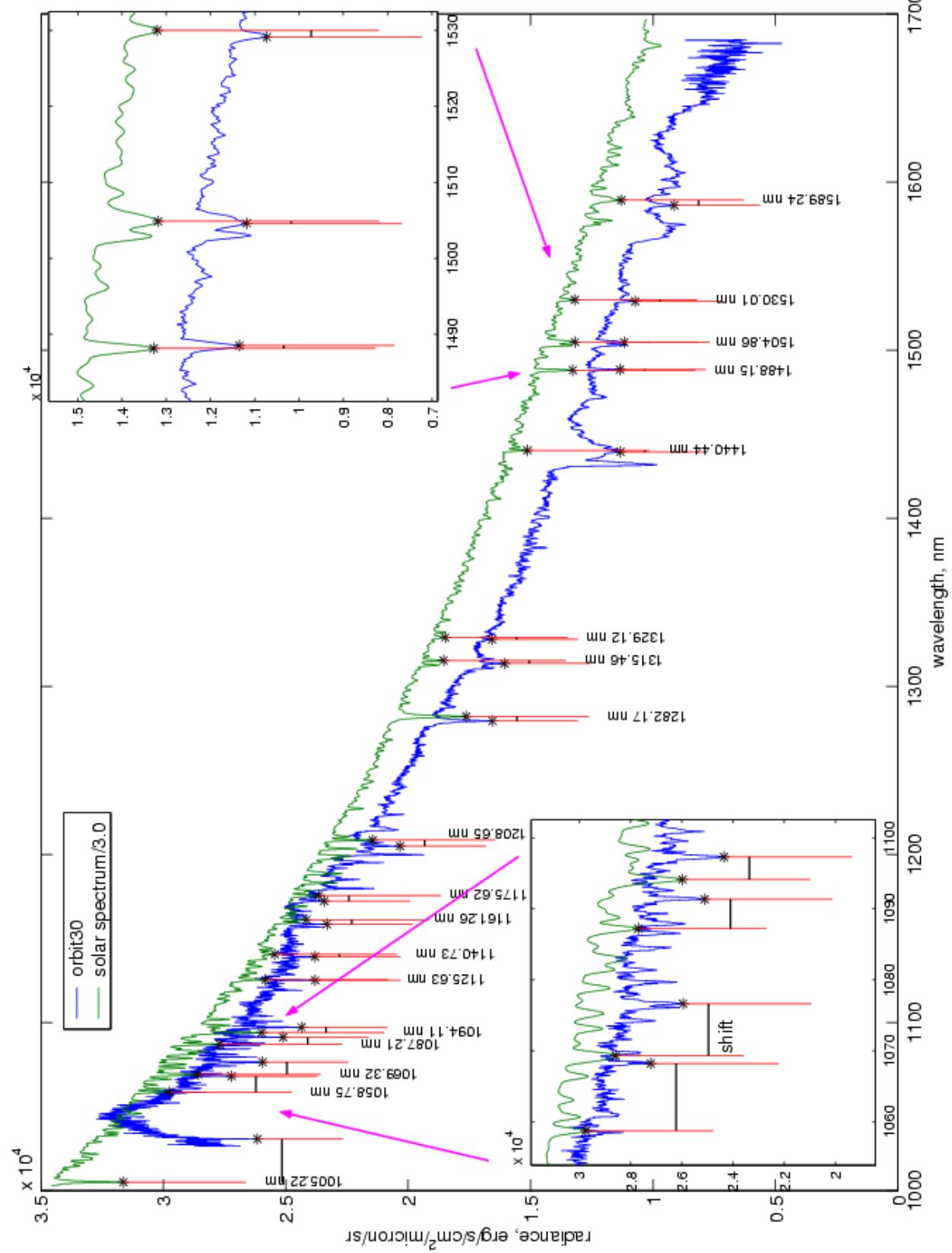
- $\lambda$  versus frequency of the AOTF depends on the crystal temperature
- Ground calibration done with HgAr pen-ray lamp for temp range  $-20^{\circ}\text{C}$  to  $+40^{\circ}\text{C}$
- Fitted with  $\lambda = a(1 + \alpha T)/f + b(1 + \beta T)$ ;  $\alpha, \beta \sim 10^{-5} \text{ K}^{-1}$  with an accuracy  $\sim 0.1 \text{ nm}$  within the entire spectral range.

## ◆ Resolving power

- Fitted to the shape of the same pen-ray lamp lines
- $\lambda/\Delta\lambda$  superior to 1700

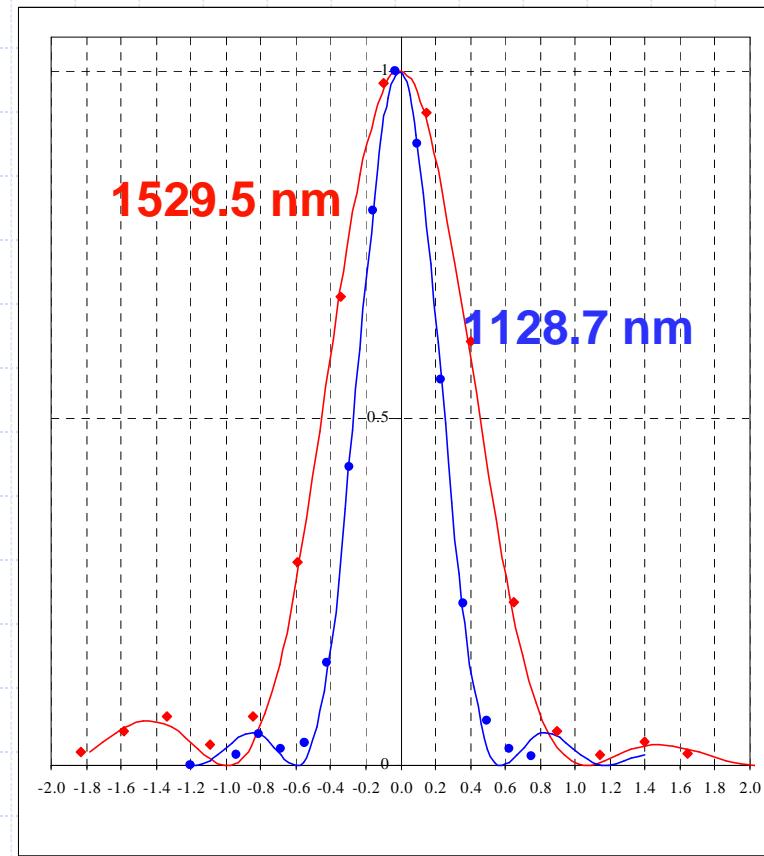
# Raw spectrum example: "full"



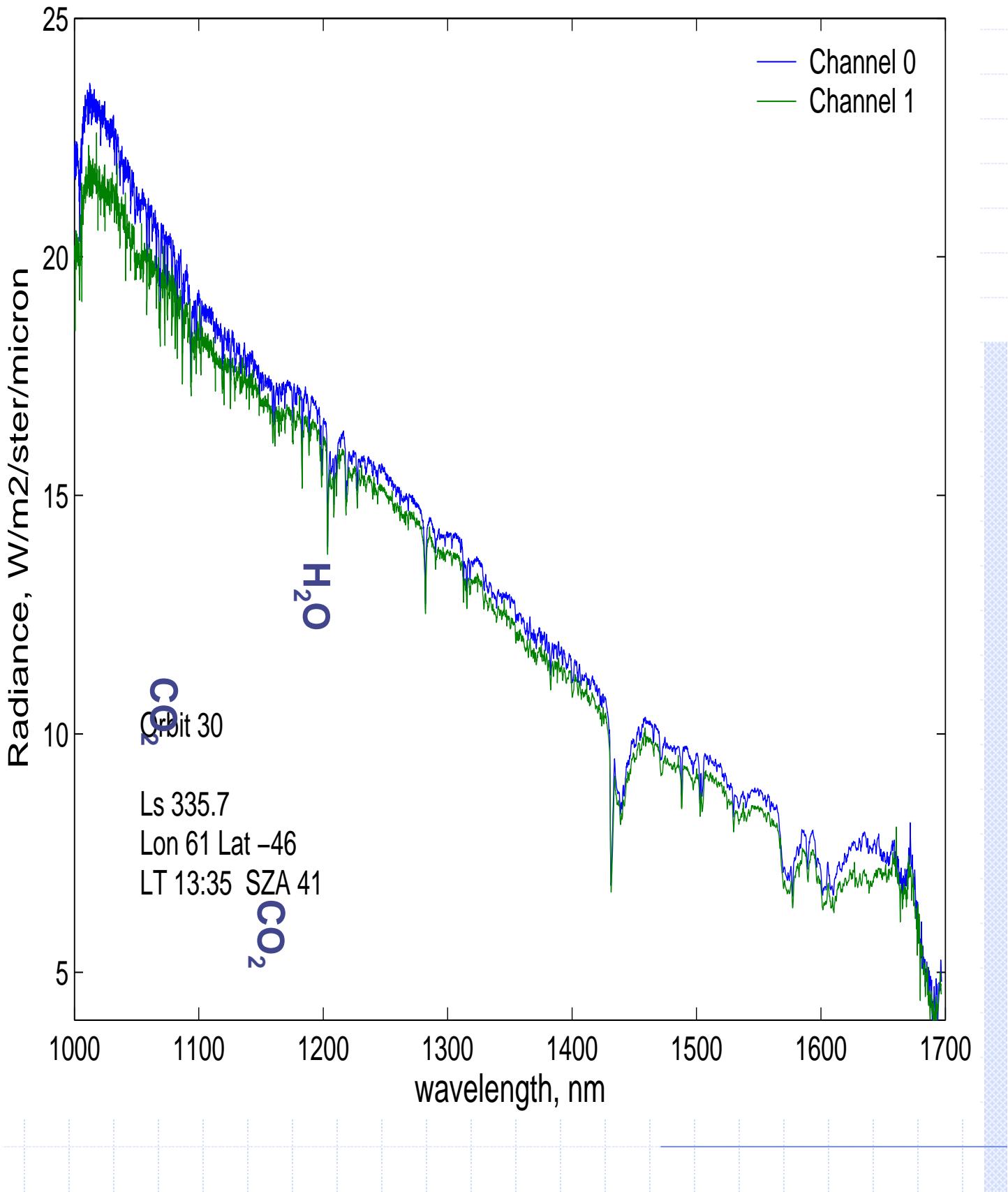


# Wavelengths calibration and resolving power

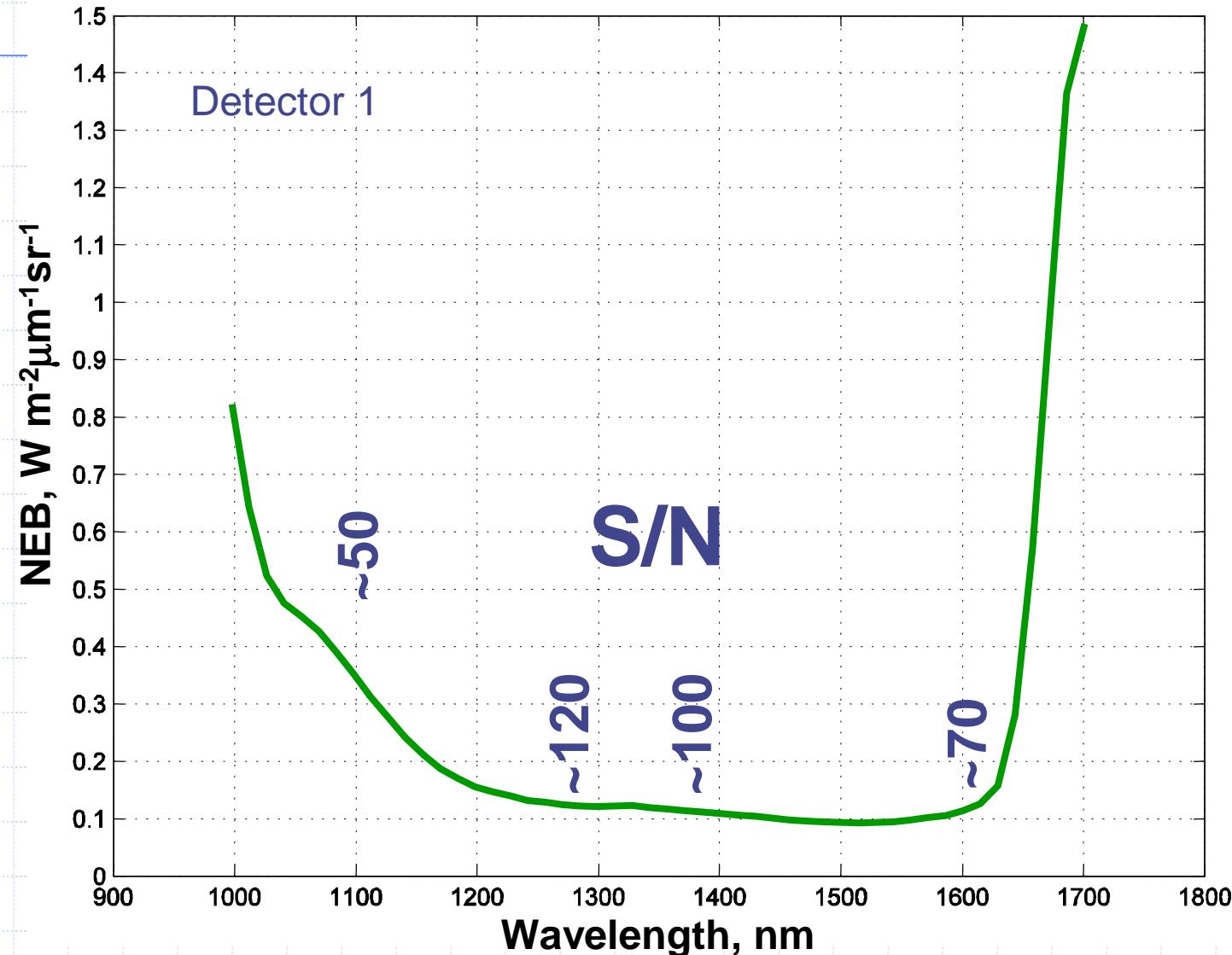
- ◆  $\lambda = a(1+\alpha T)/f + b(1+\beta T)$   
accuracy  $\sim 0.1 \text{ nm}$
- ◆  $\alpha, \beta \sim 10^{-5} \text{ K}^{-1}$
  
- ◆  $1130 \text{ nm} \rightarrow \lambda/\Delta\lambda = 2200$
- ◆  $1500 \text{ nm} \rightarrow \lambda/\Delta\lambda = 1750$
- ◆ Resolving power at least 1800 in the H<sub>2</sub>O region at  $1.38 \mu\text{m}$
  
- ◆ Sinc<sup>2</sup> $\delta\lambda \rightarrow$  tails



# Calibrated SPICAM IR data



# Noise Equivalent Brightness and S/N



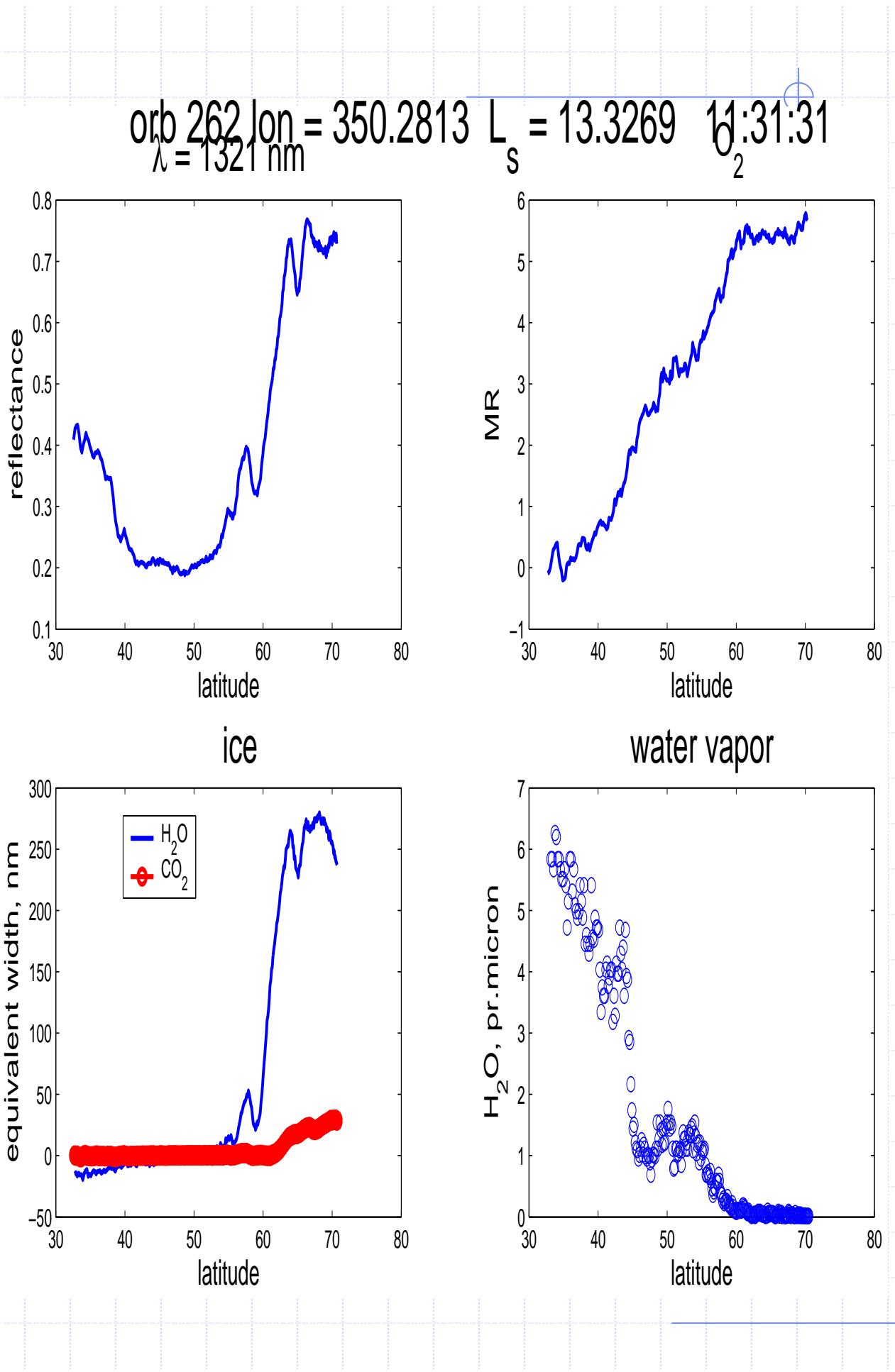
# Basic performances and most used parameters of the AOTF spectrometer

Spectral range	1-1.7 $\mu\text{m}$
Spectral resolution	0.5-1.32 nm ( $\approx 3.5 \text{ cm}^{-1}$ )
Detectors	Two InGaAs Hamamatsu photodiodes for two polarizations
FOV	$\emptyset 1^\circ$ circular
Integration time for 1 spectral point	2.8 ms or 5.6 ms
Data rate	2x332 points per second or per 2 seconds
Length of a spectrum	332 to 3984 points (multiples of 332) Always x2 polarizations
Spectral sampling (frequency step)	$\approx 0.25$ of resolution element (best sampled) to $\approx 3$ of resolution element (worst undersampled)
Time to measure 1 spectrum	1 to 24 s

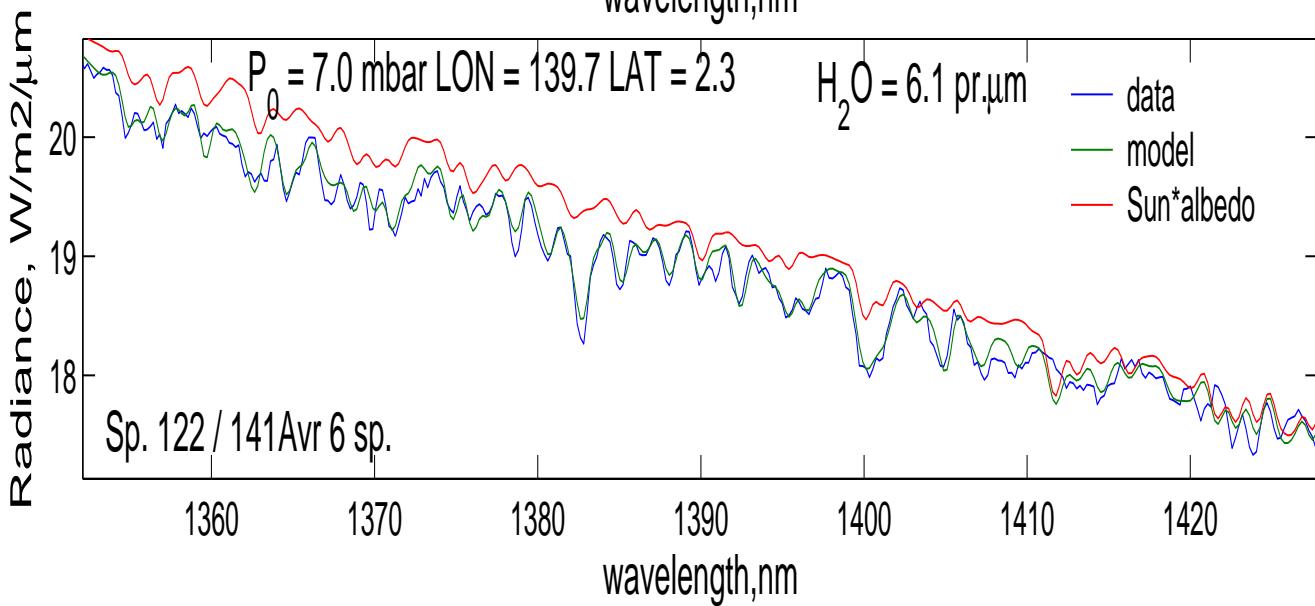
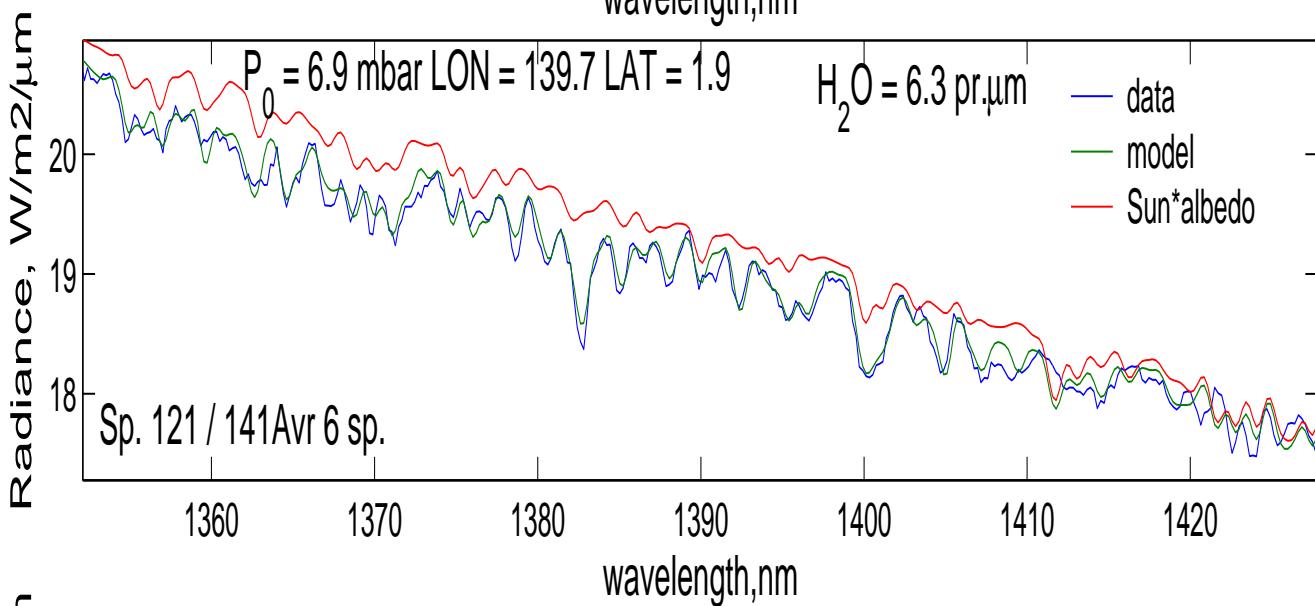
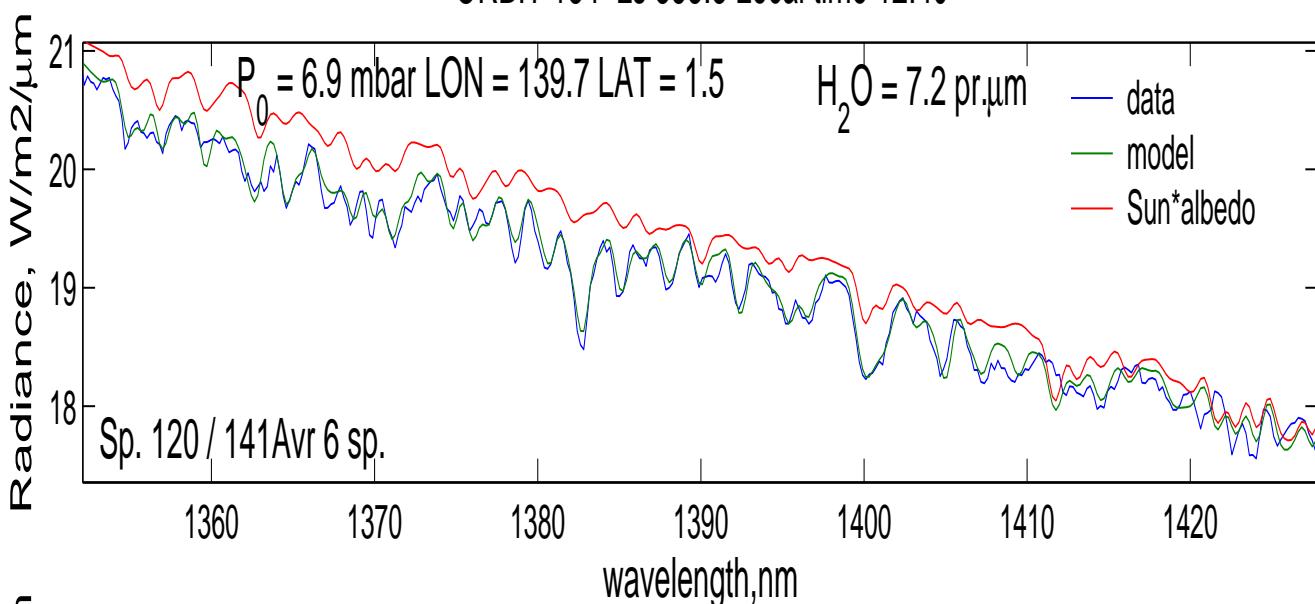
# Products of near-IR AOTF spectrometer in Nadir

- ◆ Water vapour column (pr.  $\mu\text{m}$ )
- ◆ Equivalent width of  $\text{O}_2 \text{ } ^1\Delta_g$  emission (in MR)  
proportional to  $\text{O}_3$  column above 15-20 km
- ◆ Fitting  $\text{CO}_2$  absorption at assumed surface pressure → aerosol optical depth
- ◆ Reflectance in the spectral range of 1-1.6  $\mu\text{m}$
- ◆ Equivalent width of  $\text{CO}_2$  ice absorption →  $\text{CO}_2$  ice detection (on the surface or in clouds)
- ◆ Equivalent width of  $\text{H}_2\text{O}$  ice absorption  $\text{H}_2\text{O}$  ice detection

# Some of SPICAM-IR products

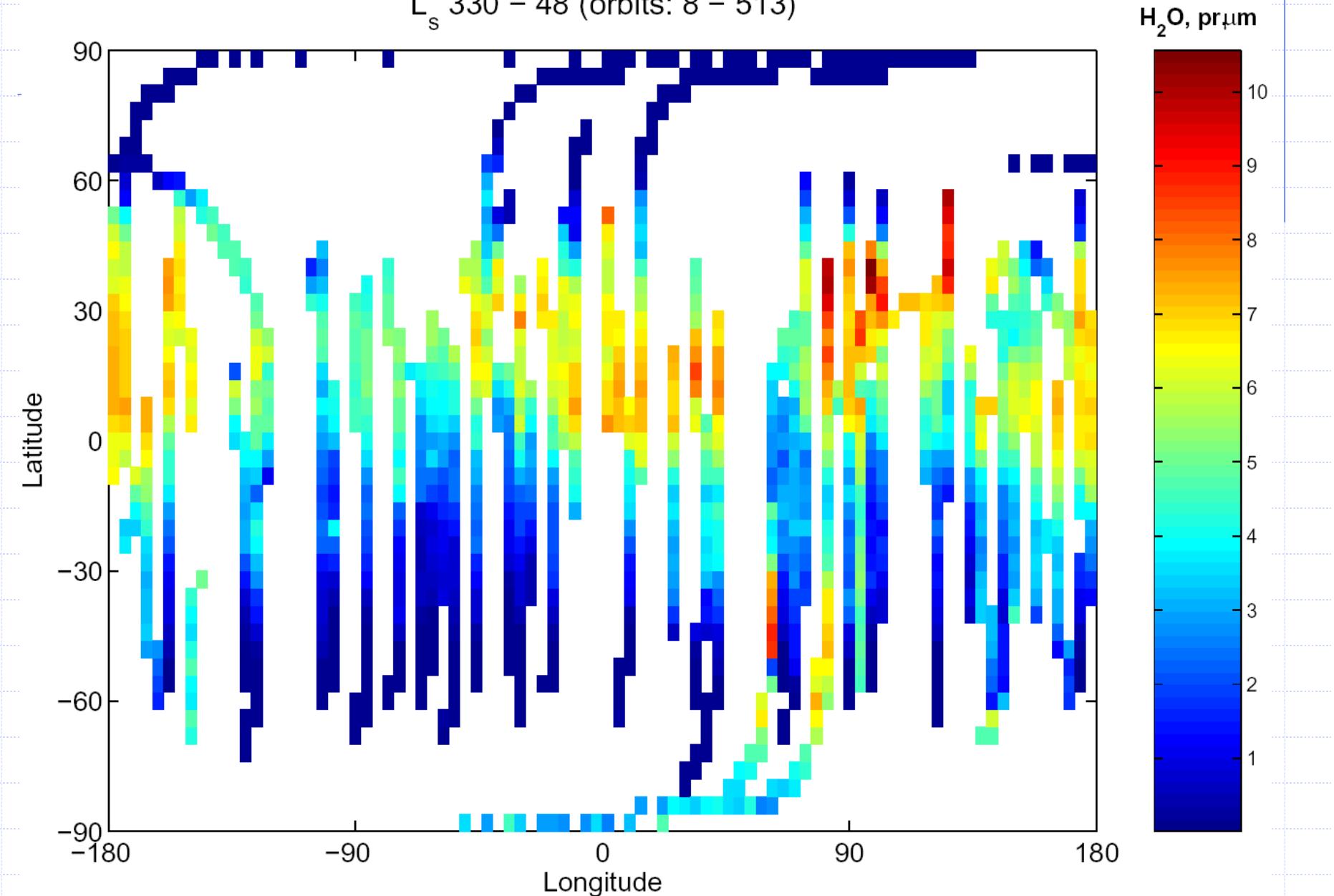


### ORBIT 134 Ls 353.3 Local time 12:40

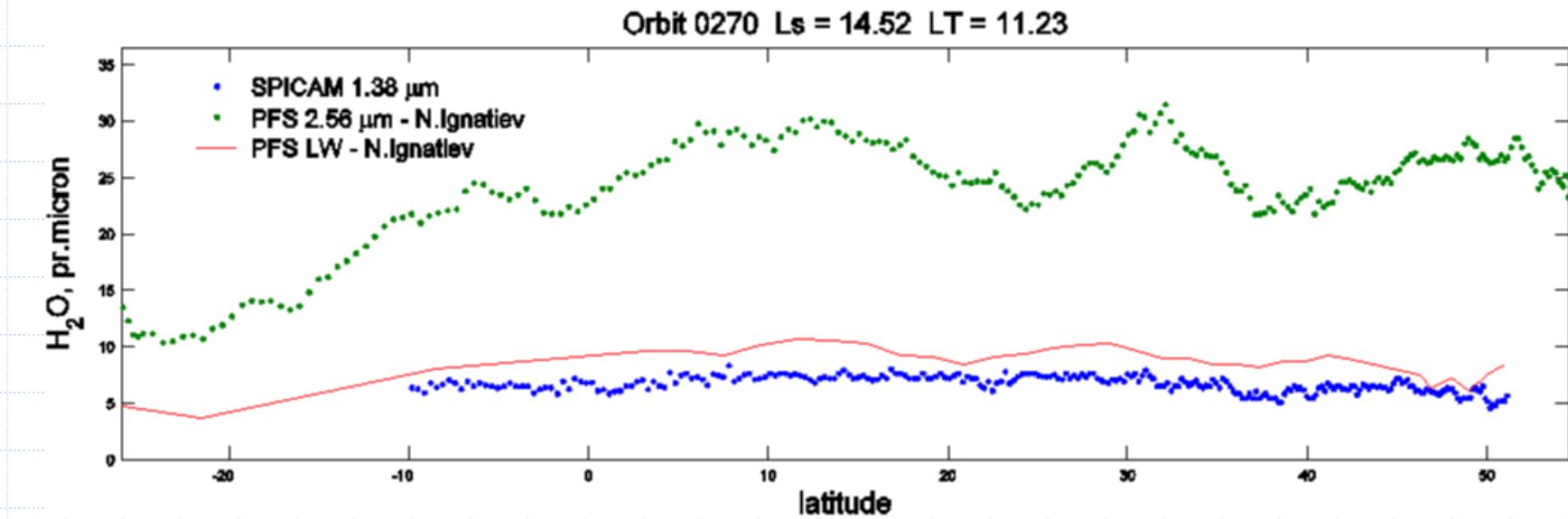
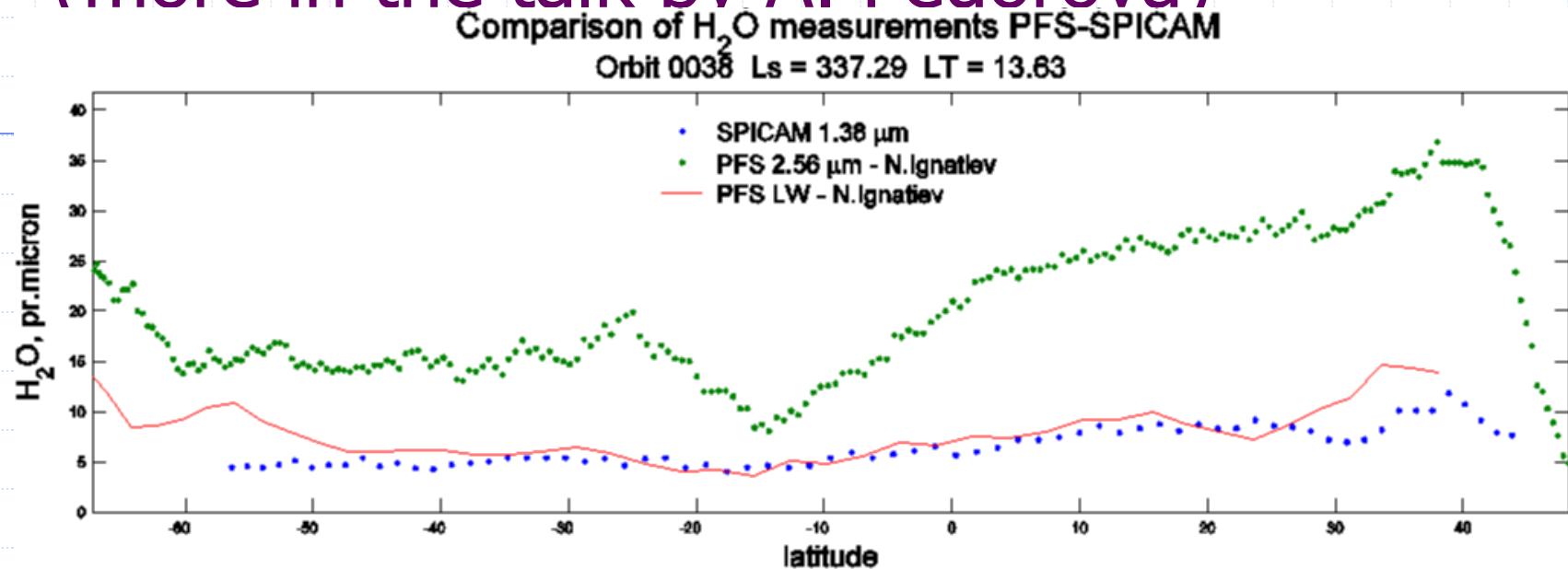


# $\text{H}_2\text{O}$ map from SPICAM measurements in $1.38 \mu\text{m}$ band

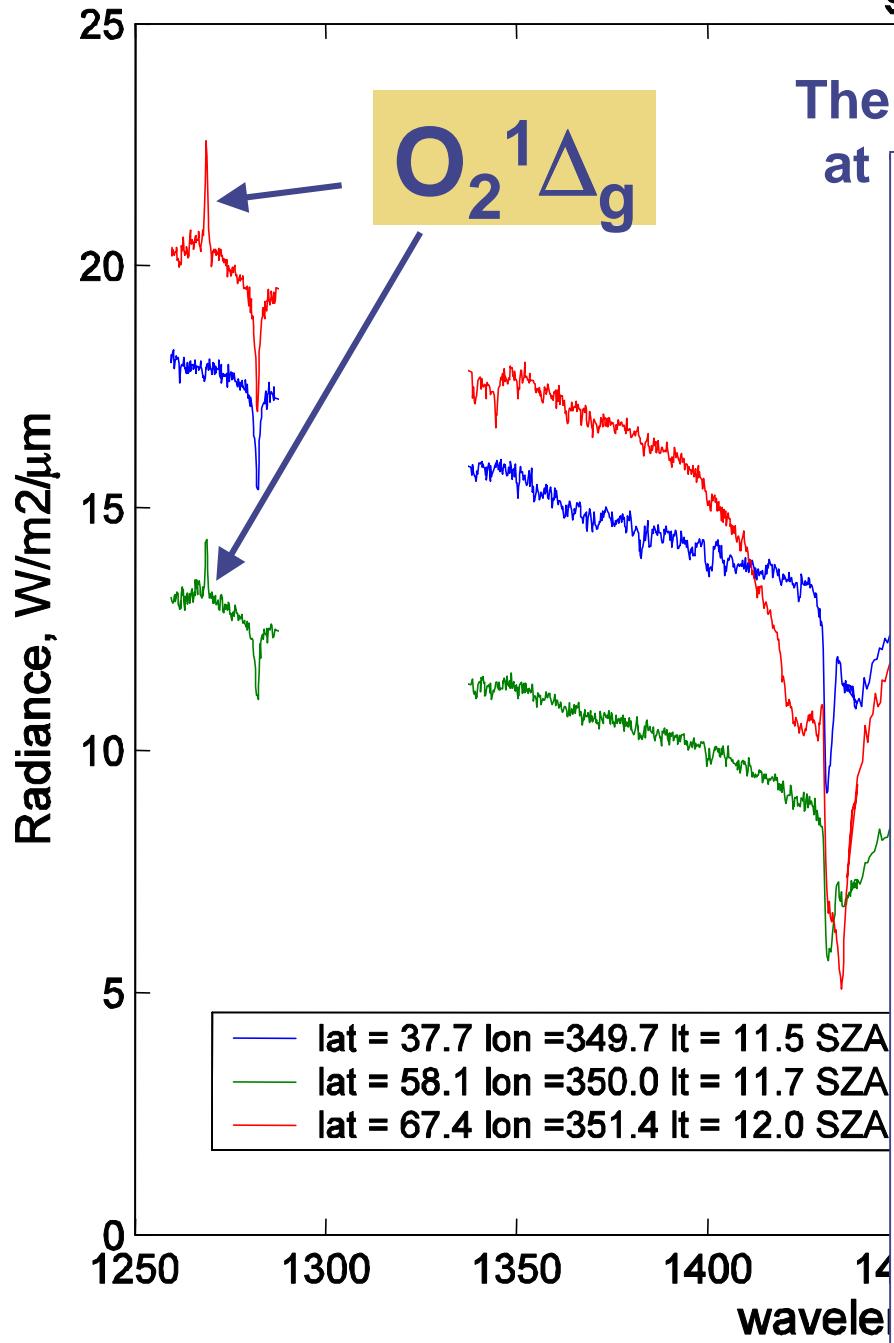
$L_s$  330 – 48 (orbits: 8 – 513)



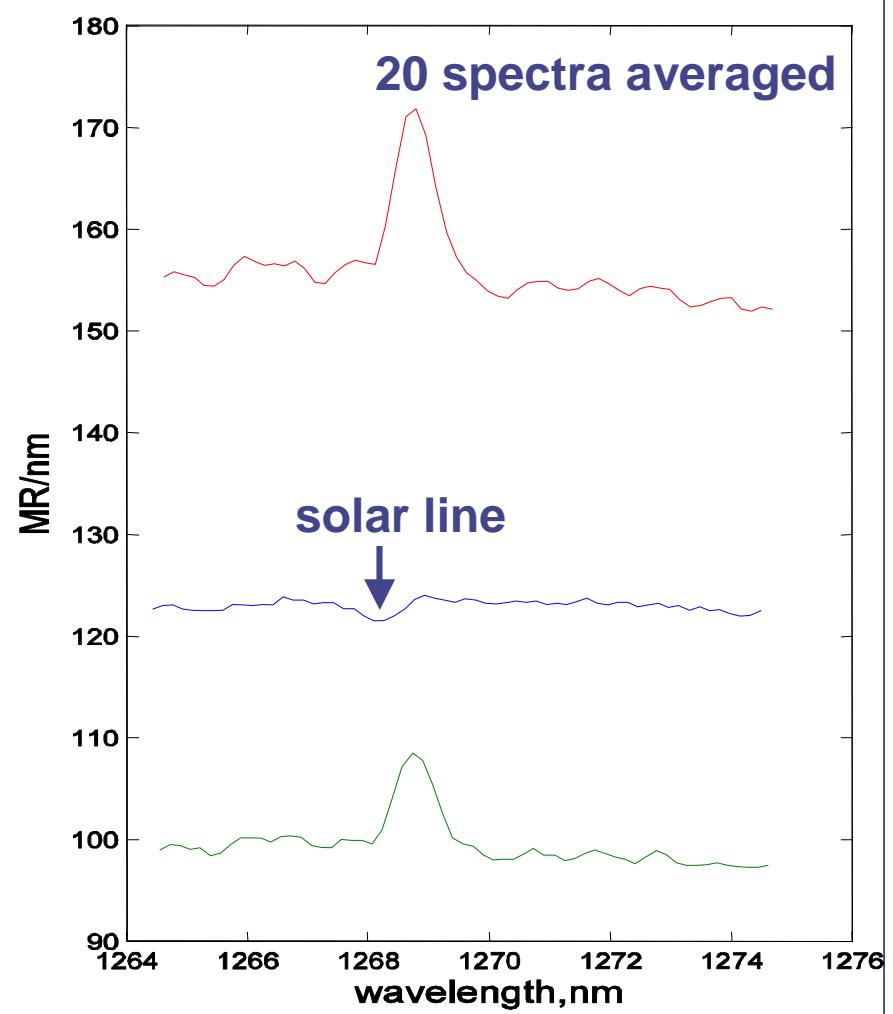
# Example of PFS-SPICAM comparison (more in the talk by A. Fedorova)

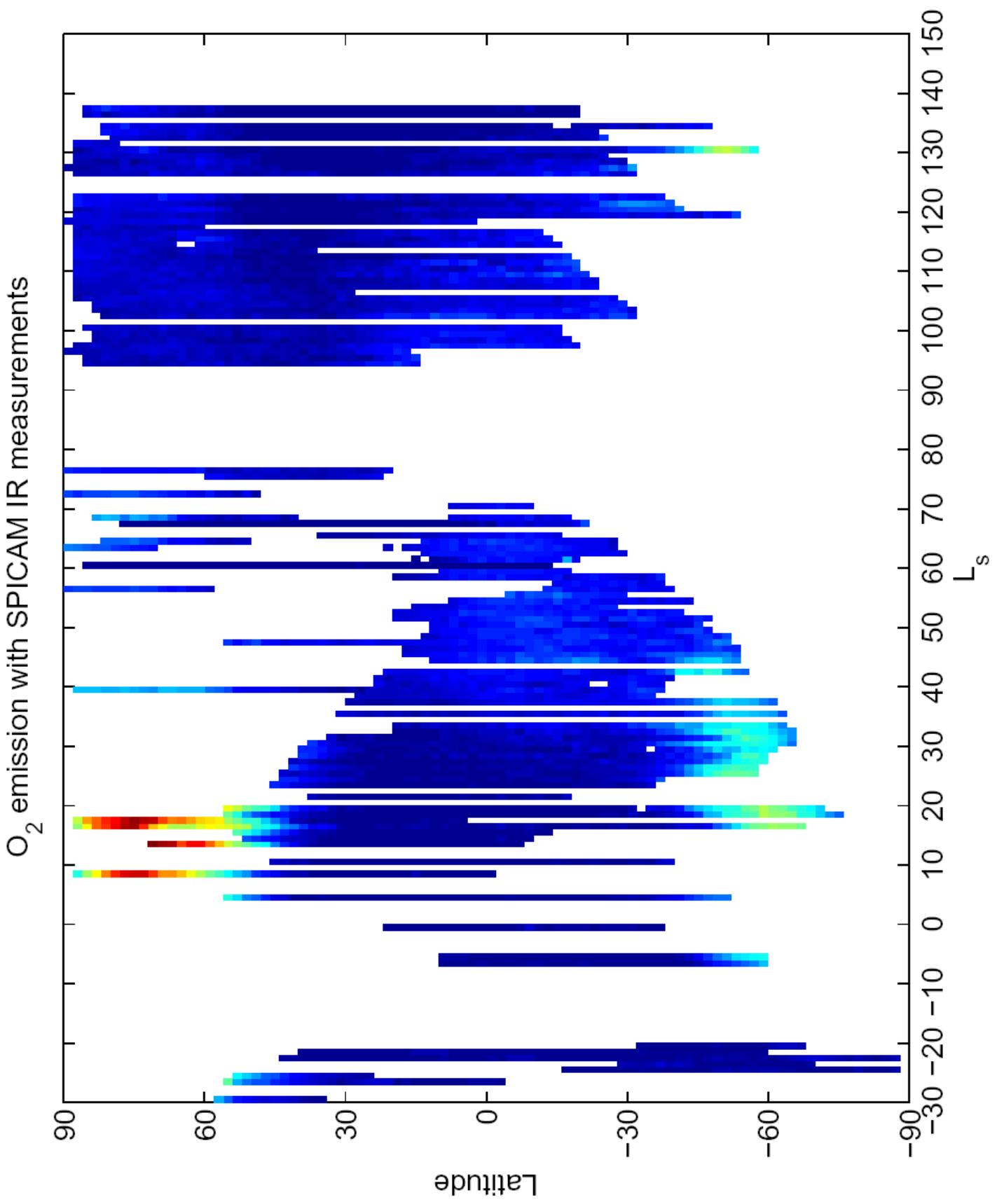


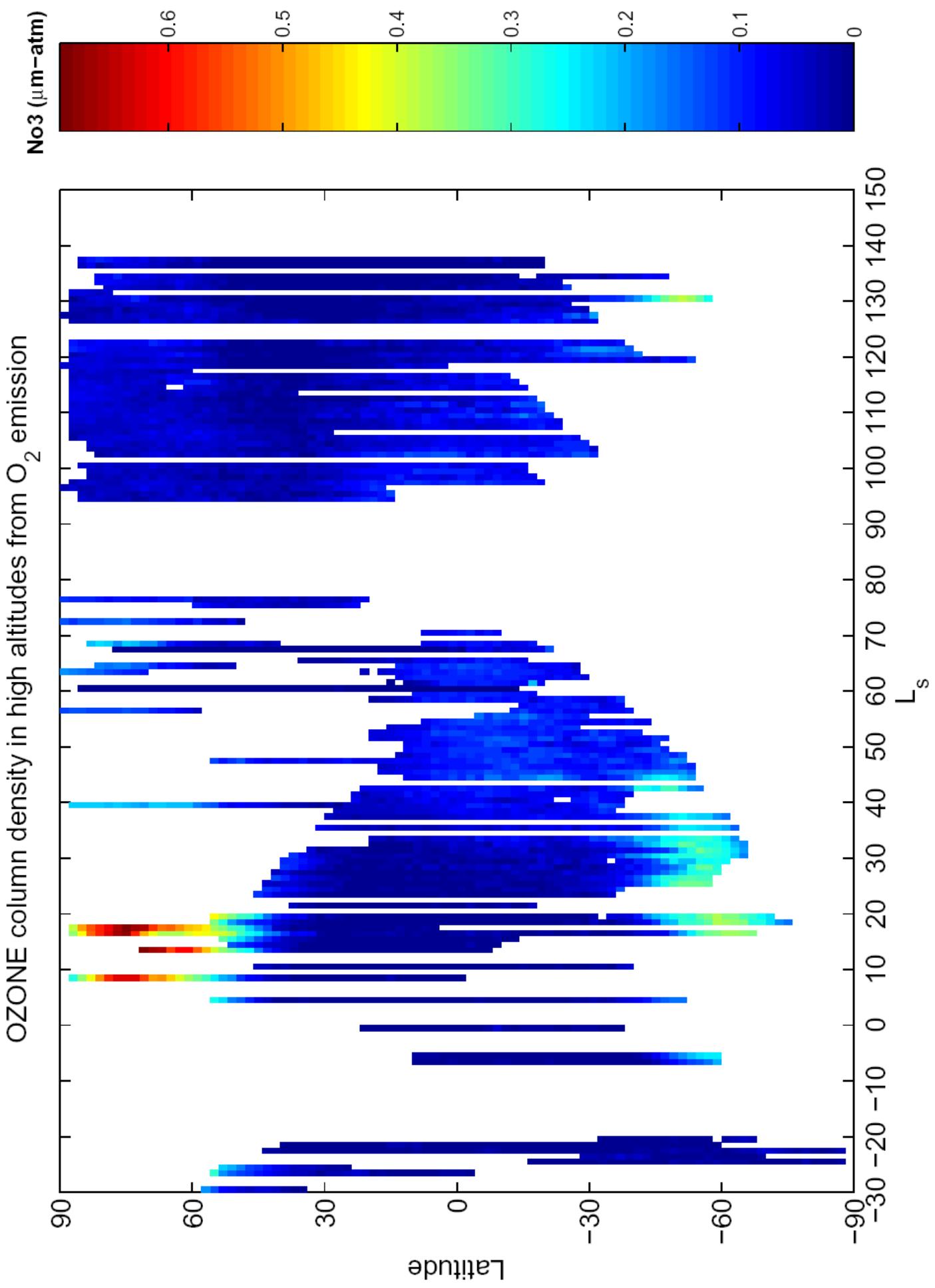
Orbit 262  $L_s = 13.3$



The oxygen  $\text{O}_2{}^1\Delta_g$  emission line  
at  $1268 \text{ nm}$

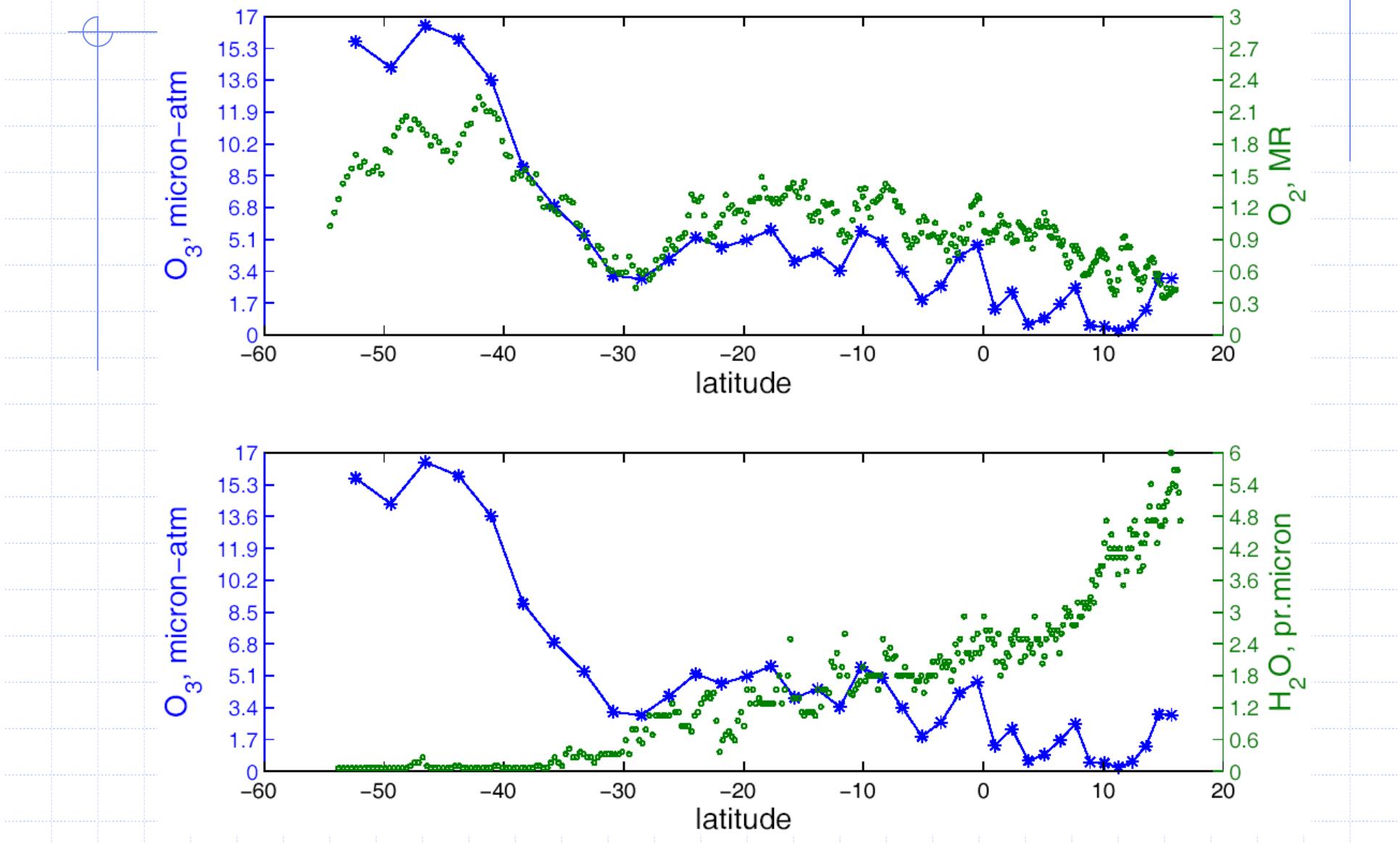






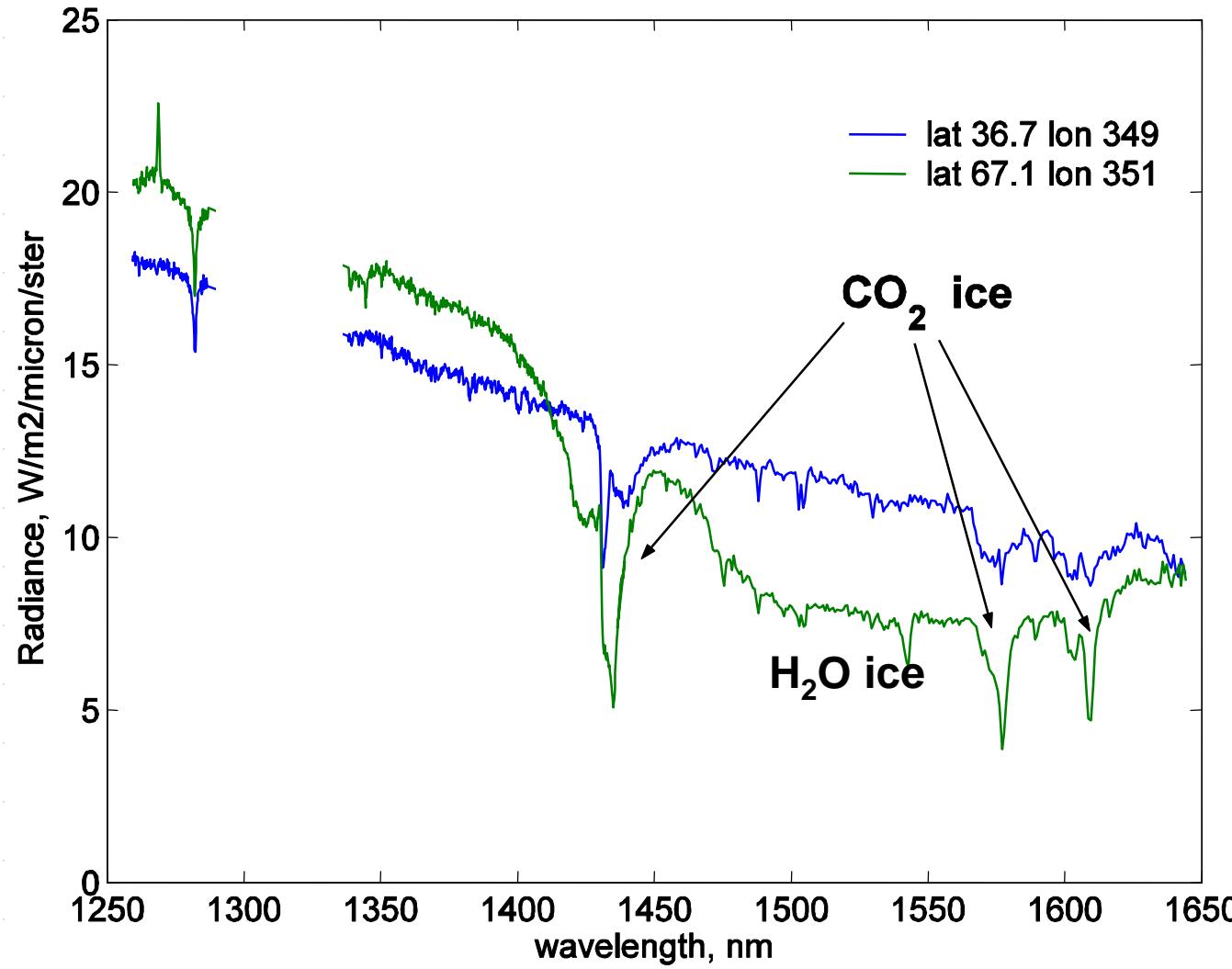
# Comparison of $H_2O$ and $O_3$ data (more in talk by S. Perrier.

Orbit 0500  $L_s = 45.9$

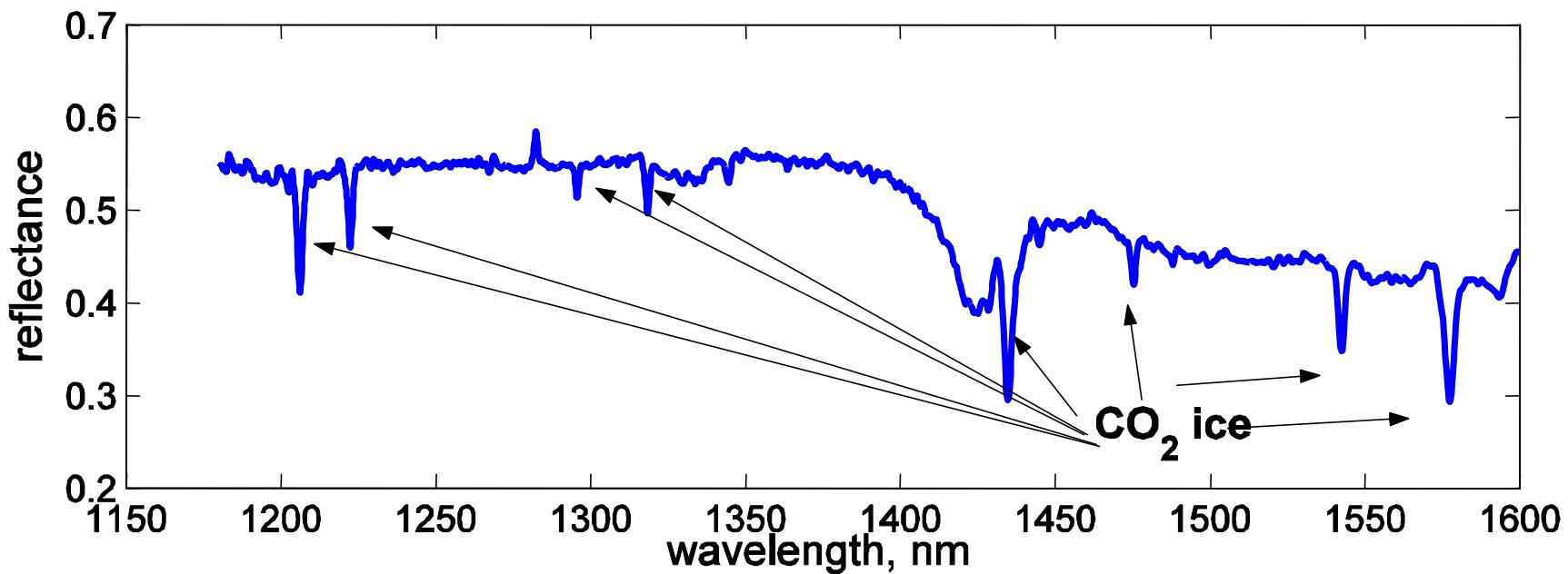
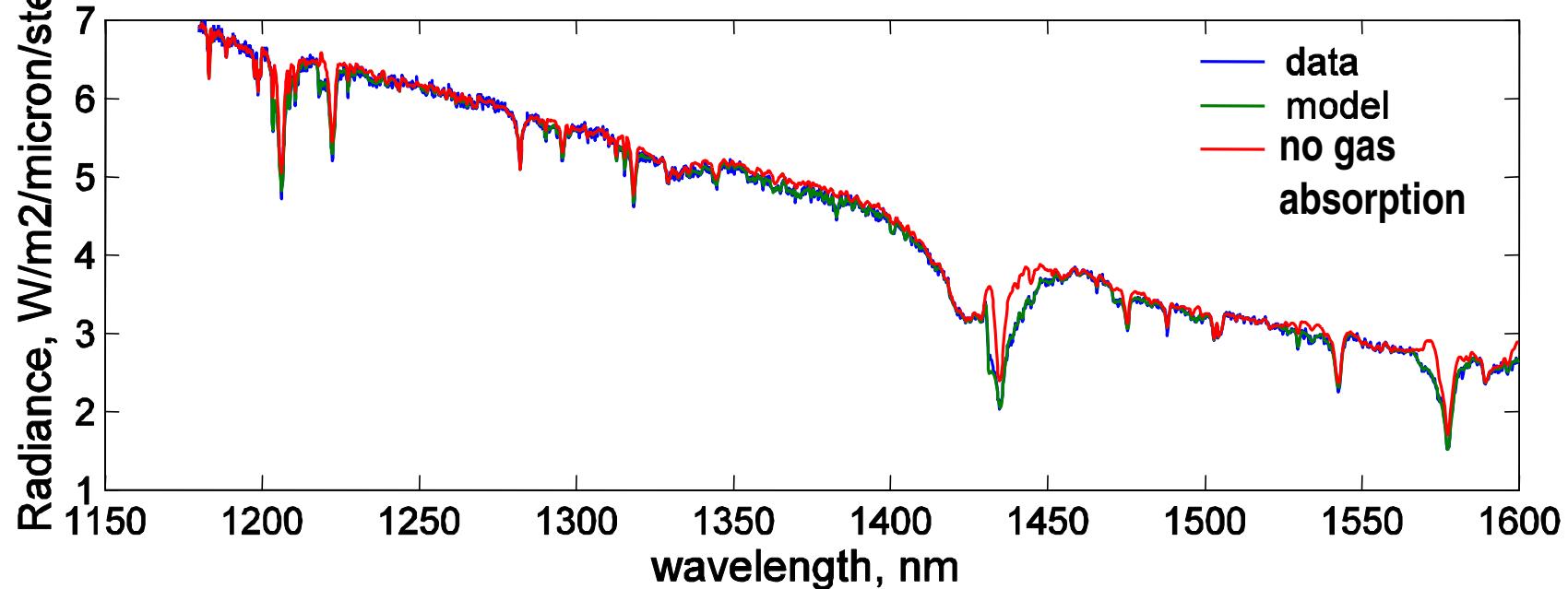


# Ice signatures: CO<sub>2</sub> and H<sub>2</sub>O

Orbit 262 Ls 13.3 LocalTime 12:00

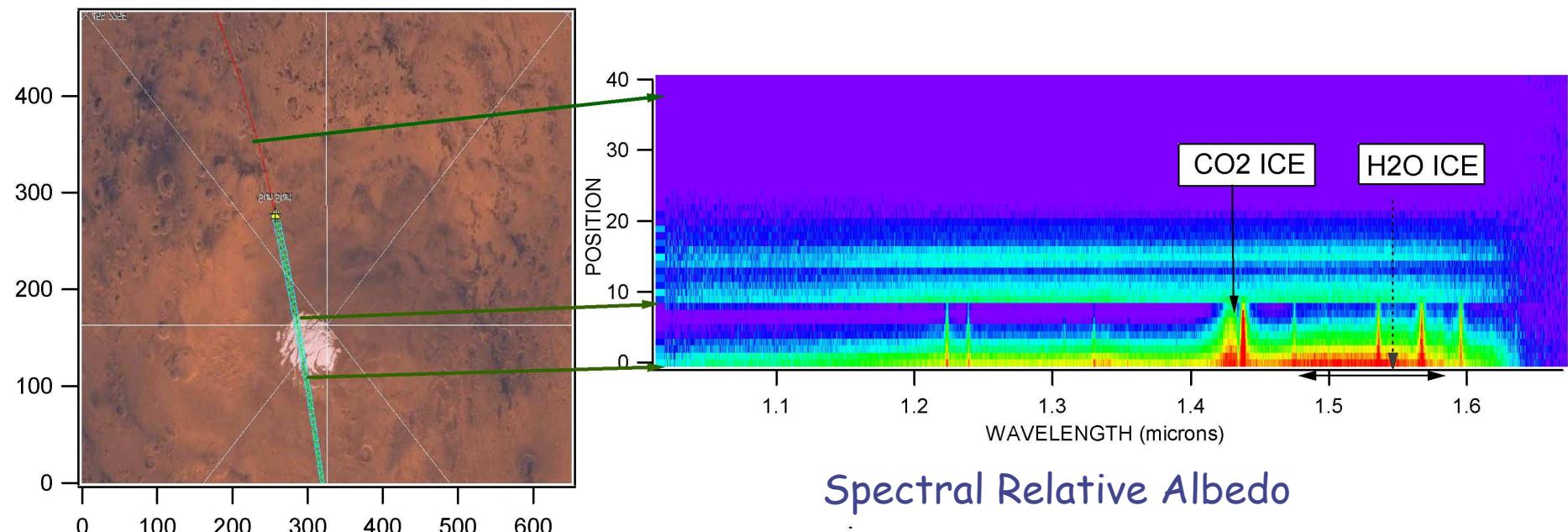


# ORBIT 30 Ls 335.7 lat -85 lon 296

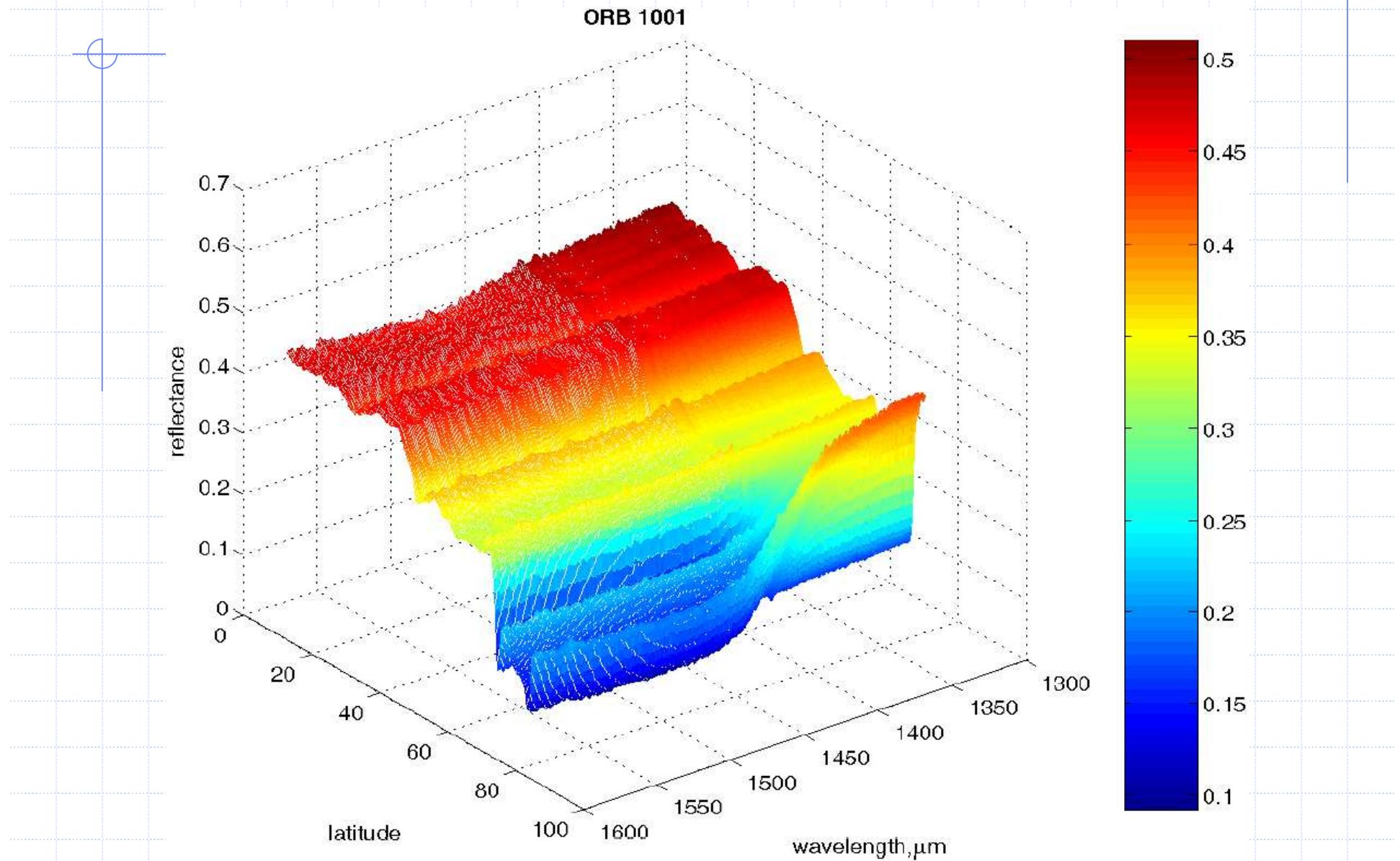


# MARS EXPRESS    SPICAM/AOTF

SOUTH POLE ORBIT 30



# Albedo in the range 1-1.7 $\mu\text{m}$



# Conclusions and plans for future

- ◆ On Mars Express SPICAM IR is to large extent duplicative to such instruments as OMEGA and PFS
- ◆ This channel proved to be capable of measuring :
  - water vapour with an accuracy of 1-2 pr. $\mu$ m
  - ozone through  $O_2^1\Delta_g$  emission
- ◆ We plan also to implement
  - routine measurements CO<sub>2</sub> and H<sub>2</sub>O ices
  - aerosols at limb,
  - Possibly, retrieve aerosol in Nadir
  - And to analyze in detail Solar Occultations (already performed)
- ◆ For the future:
  - AOTF sensor is below 1-kg threshold. S/N of SPICAM IR AOTF derivative for Venus Express has been largely improved
  - We believe that this type of instrument should be a routine versatile aeronomy-climate sensor in any future Mars missions