

# **The ESA/NASA ExoMars Programme**

## Cesa International Scene

• Recognising that a Mars Sample Return (MSR) mission is very challenging, and that its and that its undertaking will likely exceed the financial capabilities of any one agency, one agency,

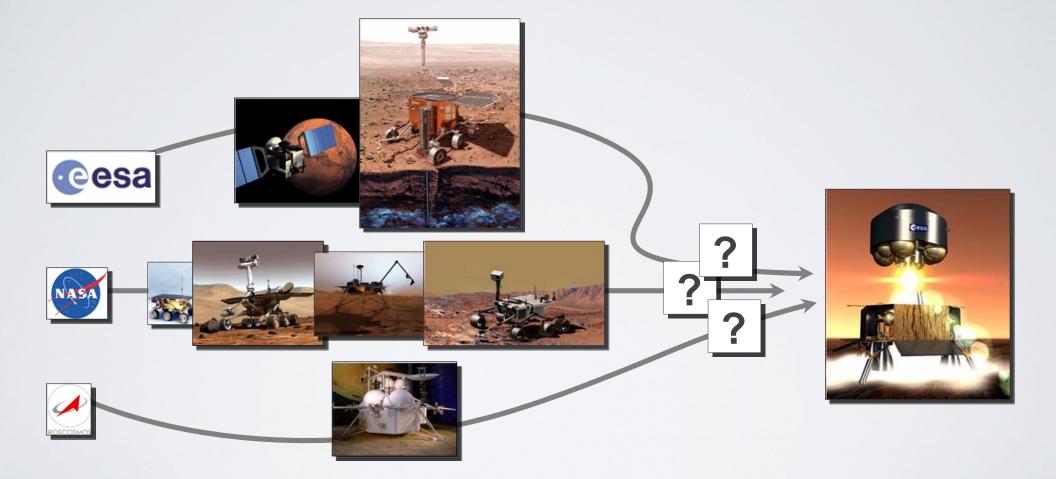
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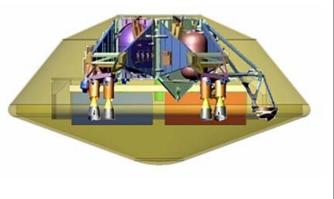
# **©esa Programme Building Blocks**

- ESA and NASA have agreed to embark on a joint Mars robotic exploration programme:
- Y Initially, seek agreement on mission configurations for 2016, 2018, and 2020 opportunities;
- $\Upsilon$  ExoMars becomes a key element of the 2016 and 2018 scenario;

 $\Upsilon$  ExoMars spreads its objectives over two opportunities.

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	2016	ESA-led mission	
	Launcher:	NASA - Atlas V 421	
	Orbiter:	ESA	
	Payload:	ESA-NASA	
· • · •	Lander:	ESA	

2018	NASA-led mission	
Launcher:	NASA - Atlas V 531	
Cruise & EDL:	NASA	(
Rover 1:	ESA	
Rover 2:	NASA	



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esa	Mission Objectives E X O M A R S			
2016	TECHNOLOGY OBJECTIVES Υ Provide data relay services to landed missions until 2022; Υ Entry, Descent, and Landing (EDL) of a payload on the surface of Mars			
SCIENTIFIC OBJECTIVE         Υ To study Martian atmospheric trace gases and their sources.				
	Methane release: Northern summe			
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#	Acronym	Description	View Modes	Observation Modes	
1	ISPECTROMETER: BROAD SHEVEN OF L		Solar Occultation only; passive radiative cooler	2 Solar Occultations per orbit (~24/day); processing interferorgrams throughout orbit	
1	SLNIR	Solar-Nadir IR Mapper: Detection and mapping of <b>specific</b> trace gases	Solar occultation; nadir and limb viewing; heat sink required (assumed to be provided by s/c)	2 solar occultations + dayside nadir/limb (60 min) on each orbit	
2			Nadir and limb, including away from velocity vector	Continuous operations switching between nadir, space, different limb; observe both sides of ground track	
2	TIR	Thermal IR profiler/mapper spectrometer or radiometer for atmospheric <b>temperature and</b> <b>dust</b> , plus <b>H2O and some</b> trace gases	Nadir and limb views, including away from velocity vector	Continuous operations switching between nadir, space, different limb; observe both sides of ground track	
2	WAC	Wide Angle Camera imaging atmospheric phenomena for discriminating between surface, dust clouds, & ice clouds	push-frame operation with .GE. 2 color bands; requires alignment with ground track motion	Cross-track (nearly orthogonal to velocity vector) horizon-to-horizon	
3	HRCSC	High Resolution Color Stereo Camera: <b>Surface imaging</b>	~1 m/pixel ground sampling (at nadir) with TDI; fore/nadir/aft views	Designated targets of opportunity; requires alignment with ground track motion (mitigation needed)	

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#### **TECHNOLOGY OBJECTIVES**

YSurface mobility with a rover (having several kilometres range);

2018

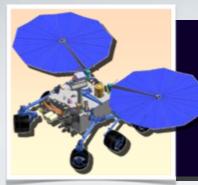
YAccess to the subsurface to acquire samples (with a drill, down to 2-m depth); YSample acquisition, preparation, distribution, and analysis.

#### SCIENTIFIC OBJECTIVES

YTo search for signs of past and present life on Mars;YTo characterise the water/subsurface environment as a function of depth in the shallow in the shallow subsurface.

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### OBJECTIVE

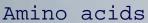
To identify, acquire, document, and cache "outstanding" samples in a manner suitable manner suitable for collection by a future Mars Sample Return mission.



# What to Search For

• **PRESENT LIFE:** Biological markers, such as:

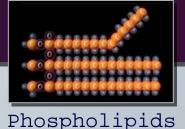






#### ds Nucleobases

#### Sugars



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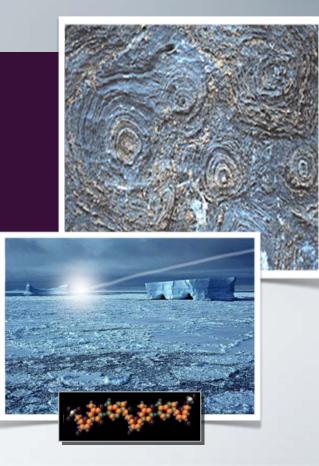


Pigments

 PAST LIFE: Organic residues of biological origin; (chemical, chiral, spectroscopic, and isotopic info)
 Images of fossil organisms and their structures; (morphological evidence)

# • DELIVERED ORGANICS:

by meteoritic and cometary infall



# **The EDLS Demonstrator**



- Maximum Mass at launch 600Kg
- Semi-soft (almost soft) lander with liquid retro-propulsion
- Landing gears (TBD) crushable material, vented airbabags, legs
- Minimum Payload operated on primary battery (5-7 sol lifetime)
- Mass less than 5Kg and no deployment mechanisms
- Science Goals: mostly environmental assessment



Nominal mission: Nominal science:

EC length: Rover mass: Mobility range: 180 sols
6 Experiment Cycles +
2 Vertical Surveys
16 – 18 sols
300 kg
Several km

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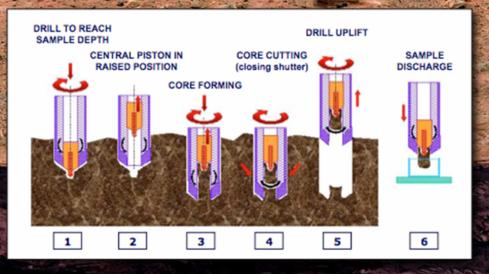
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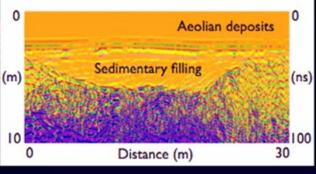
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2-m depth



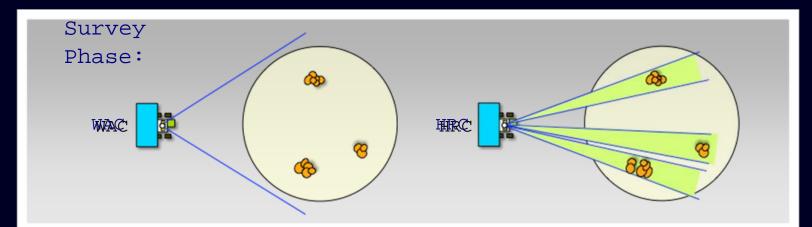
Ground Penetrating Radar 0 (m)



## Cesa Site Characterisation

# **AT PANORAMIC SCALE:** To establish the geological

context



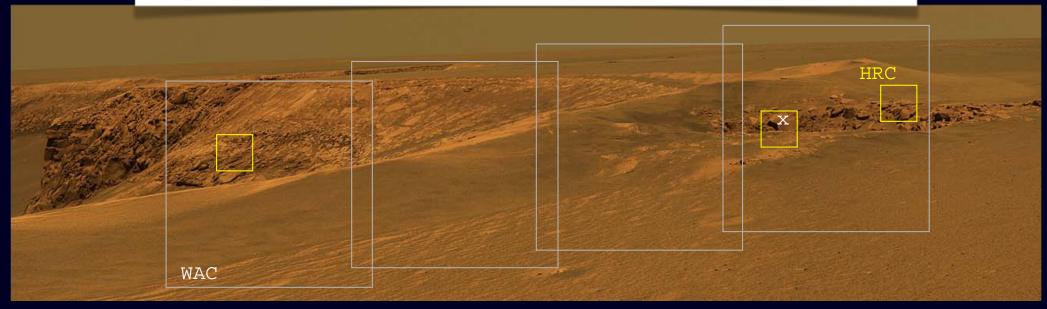
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## Cesa Outcrop Characterisation

## **AT ROCK SCALE:** To ascertain the past presence of

water

For a more detailed morphological

examination



High-Resolution

Camera

Close-Up Imager



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Next step: ANALYSIS	
Use the drill to collect 🥜	From an outcrop
a sample	From the
	subsurface



Subsurface Drill

#### OBTAIN SAMPLES FOR ANALYSIS:

#### From 0 down to 2-m depth

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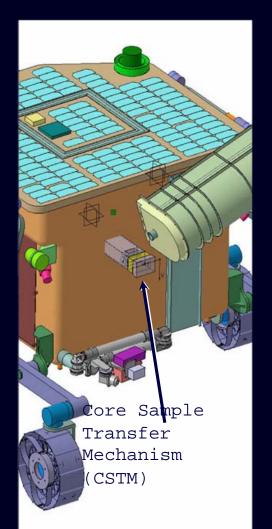
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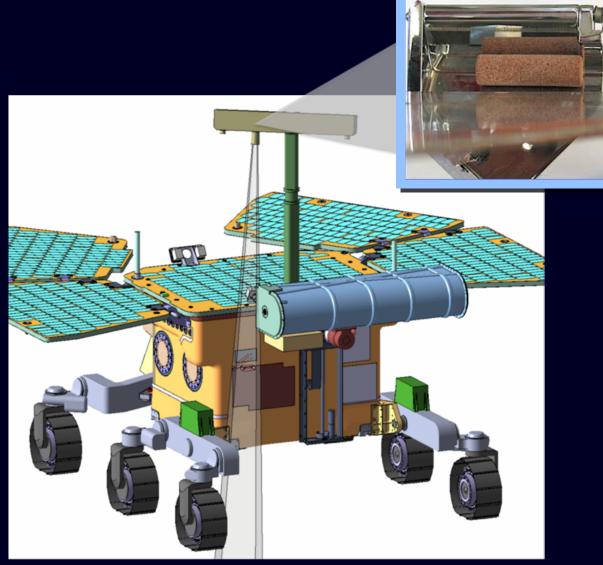
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DRILL discharges sample into Core Sample Transport Mechanism (CTSM). PanCam HRC images sample.



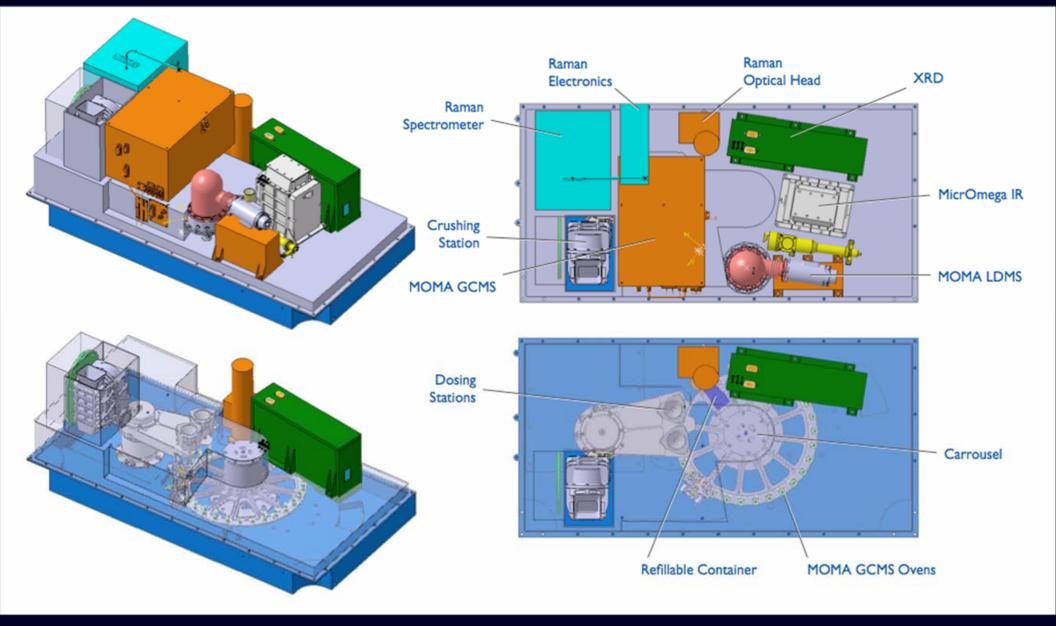


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# Cesa Analytical Laboratory



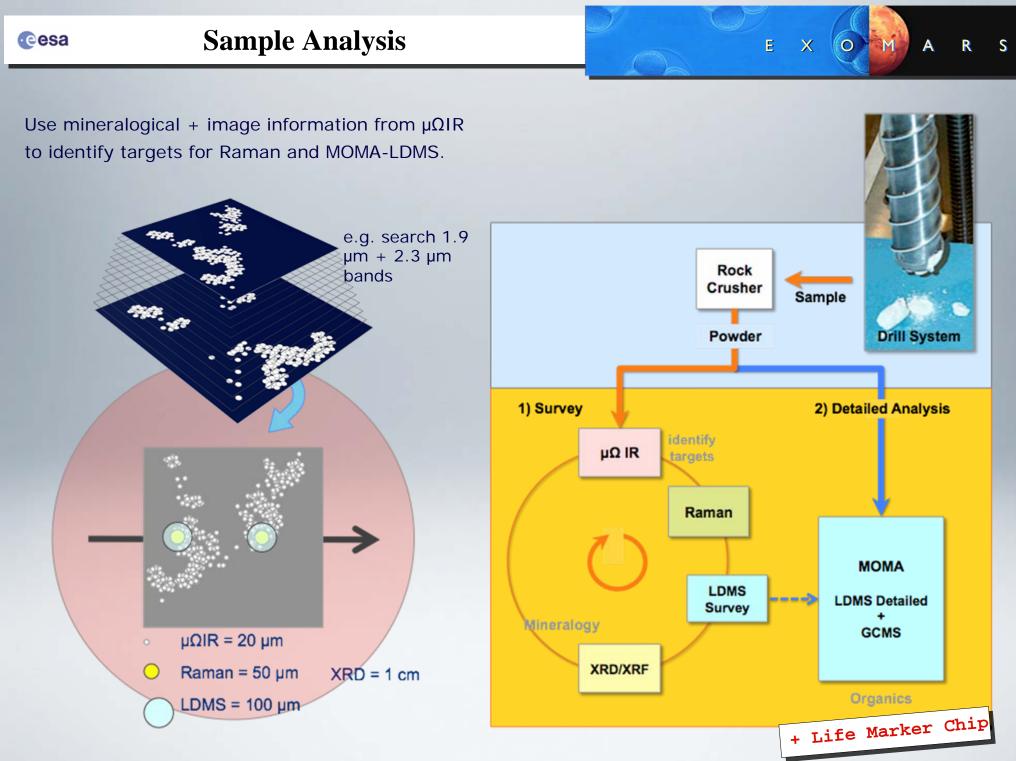
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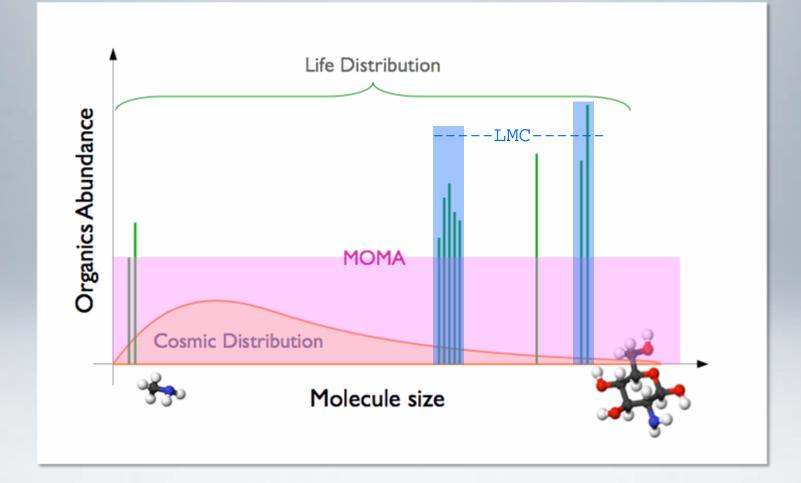
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# **Sample Analysis**



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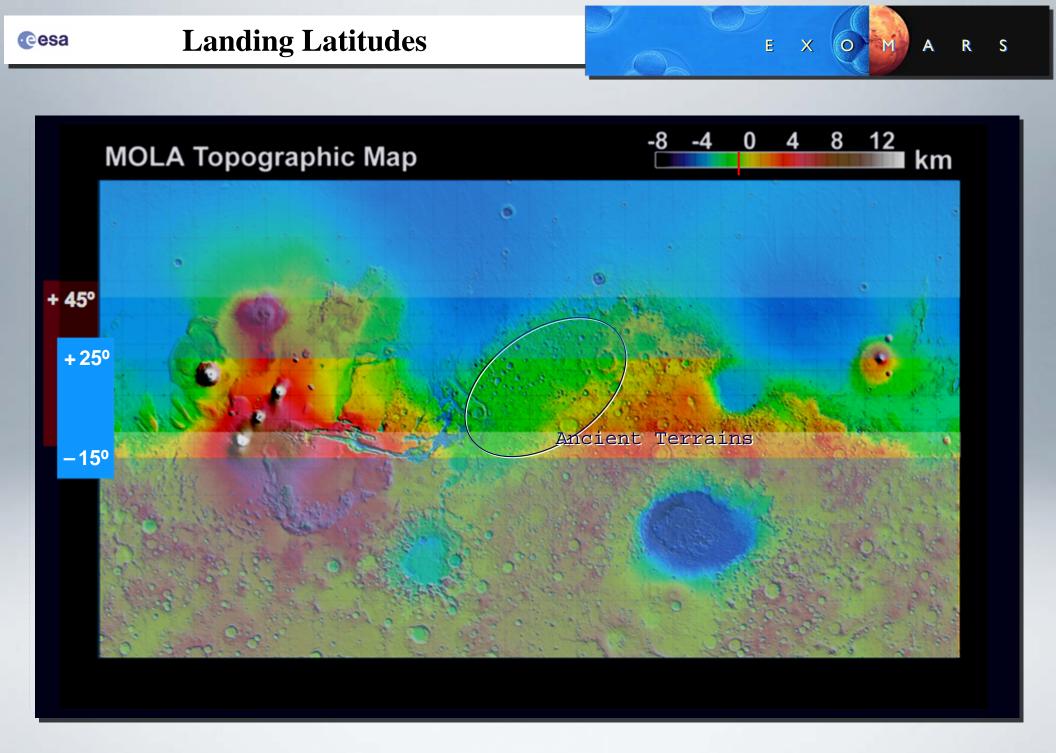
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# CesaApproved Pasteur Payload

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Instrument Name	Description	Mass (kg) including maturity margin	
PanCam (WAC + HRC)	Panoramic camera system	1.560	
MOMA	LD-MS + Pyr GC-MS for organic molecule characterisation	6.100	
MicrOmega IR	IR imaging spectrometer	0.960	
Mars-XRD	X-ray diffractometer + X-ray fluorescence	1.480	
Raman (internal)	Raman spectrometer	2.260	
WISDOM	Shallow ground- penetrating radar	1.380	
Ma_Miss included in 2.0-m drill	IR borehole spectrometer + CI	LUPI	
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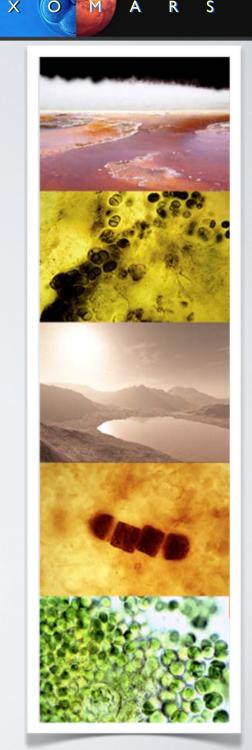


# Cesa Sites' Scientific Interest

**ExoMars is a powerful exobiology mission:** 

We will search for signs of life... in places where nobody has looked, with instruments that no one has yet flown.

- The proposed landing sites must cater to this overarching scientific goal. scientific goal.
- Proposed landing sites must contain evidence strongly suggestive of a past or of a past or present habitable environment.
- The prevailing <u>outcrop mineralogy</u> must be favourable for the long-term long-term preservation of organic molecules.
- Recently exhumed locations are particularly attractive —because of the longof the long-term radiation shielding provided while they were buried. buried.
- Proposed sites must include several attractive targets within the ellipse.
- Proposed sites must maximise opportunities for useful subsurface science science investigations.
  - ... and be safe for landing.



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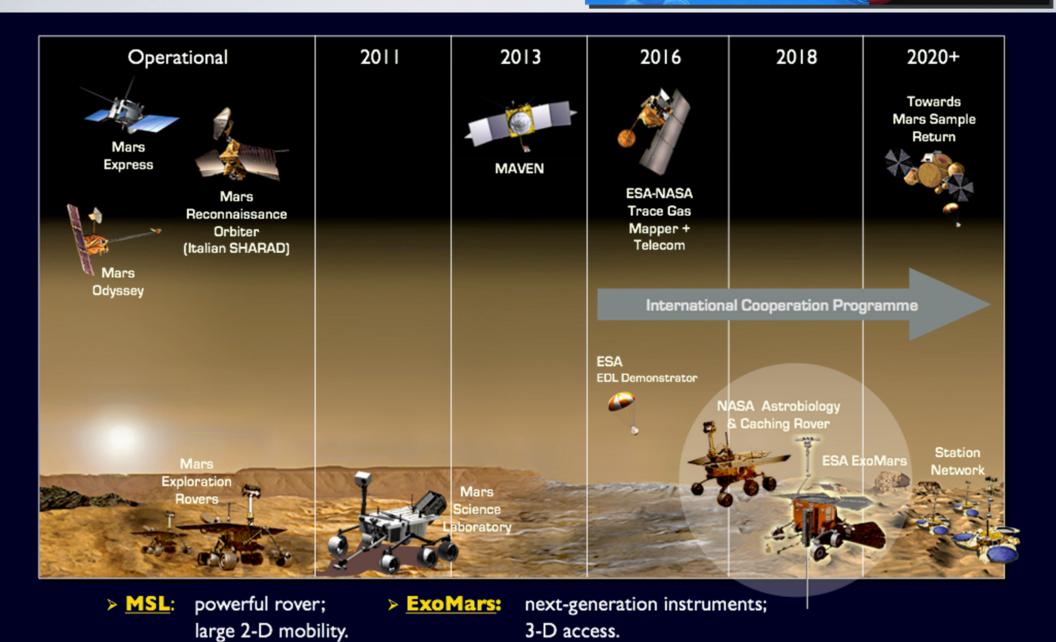
**International Scenario** 

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Following on the results of MSL, the ExoMars Rover is the logical next step in Mars surface exploration.