

**ESA Science & Technology** 

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# No. 51 - Second Push-Broom Operations Phase

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Report for period 17 April to 14 may 2006

The completion of the thermal analysis being done as part of the preparation for the SMART-1 Moon impact gave the Flight Control Team a surprise. The expected increase of temperatures during May due to the Sun/Earth/Moon alignment was going to be higher than six months ago.

This especially affects the solar arrays at perilune when SMART-1 passes the subsolar point with the Sun on one side and the illumination by light reflected of the full Moon on the other, causing temperatures to rise above 105 °C. This would be close to the qualification limit of the solar arrays of 110 °C.

After several considerations including the trend analysis of the real time telemetry, it was decided not to take any risk and implement a 35° offset in the solar array orientation, enough to keep it within a reasonable range of temperatures. This was implemented on 9 May 2006 and proved to be successful, lowering the maximum reached temperature to around 85 °C instead of the 96 °C prior to the implementation of the offset.

Other SMART-1 operations have been nominal during this period which saw the start of the second push-broom observations phase.

Moon impact preparation:

- Flight Dynamics strategy is completed
- The Swedish Space Corporation Analysis is completed
- The Thermal analysis and simulations are completed
- The new procedures development, testing and validation are completed
- The SMART-1 Principal Investigator, Bernard Foing is finalizing the coordination of the SMART-1 Moon impact and observation campaign

The SMART-1 Ground Operations Automation System (GOAS) will be used operationally in the second half of May, which is a first for an ESA mission.

## **Future Activities**

The future activities are focused on the following:

- Procedures development to raise the orbit by 90 km using the attitude thrusters in June/July
- Finalise documentation and release of the end of mission and Moon impact analysis
- Tone ranging test on SMART-1 with Chinese CLTC Kashi station and with ISRO
- Start using the SMART-1 Ground Operations Automation System (S1 GOAS)
- Preparation of papers for Spaceops and IAA in Valencia

Note: The second ranging test with the Chinese station Kashi is now confirmed for week 22.

#### Spacecraft Status

AOCS

The AOC subsystem has done well in the reporting period. The solar array off-pointing and the push-broom activities did not affect the overall performance of the AOCS.

## **Electric Propulsion, Power and Thermal**

## **Electric Propulsion**

The Electric Propulsion has been off during the reporting period.

## Power

The reporting period saw the start of an eclipse season. The evolution of the eclipse duration can be seen in the plot.



Eclipse duration in hh:mm:ss (total for Umbra and Penumbra)

During the reporting period the maximum eclipse occurred on 14 May and lasted just over one hour:

#### Time (UT) Event

22:02:53	SMART-1 enters Moon Penumbra
22:03:23	SMART-1 enters Moon Umbra
23:02:45	SMART-1 leaves Moon Umbra
23:03:26	SMART-1 leaves Moon Penumbra

The performance of the Power Subsystem was very good. As part of the routine analysis during the eclipse, a check of the battery cell capacity was done, showing that there is no degradation of the cells and that the performance is as expected, when compared with previous eclipses.

#### Thermal

The Thermal Subsystem has performed very well during the reporting period.

## **Orbital Information**

SMART-1 OD432 Close to Apolune 2363 Epoch (UTC) 2006/05/15 09:50:25.0

#### Elements WRT Moon and its equator of date

Pericentre Distance (km)	2133.593201
Apocentre Distance (km)	4703.495591
Semi Major Axis (km)	3418.544396
Eccentricity	0.375877
Inclination (°)	90.852895
Ascending Node (°)	239.345651
Argument of Pericentre (°)	232.698778
True Anomaly (°)	180.000000
Osculating Orbital Period (h)	4.982153

The changes in four parameters since apolune 2329 are:

- semi-major axis -0.3 kmperilune height -36.3 km
- apolune height +35.7 km
- orbital period -0.0 min

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